

Article

# Influence of Powder Surface Contamination in the Ni-Based Superalloy Alloy718 Fabricated by Selective Laser Melting and Hot Isostatic Pressing

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Received: 9 May 2017; Accepted: 7 September 2017; Published: 13 September 2017

**Abstract:** The aim of this study was to gain a deep understanding of the microstructure-mechanical relationship between solid-state sintering and full-melting processes. The IN718 superalloy was fabricated by hot isostatic pressing (HIP) and selective laser melting (SLM). Continuous precipitates were clearly localized along the prior particle boundary (PPB) in the HIP materials, while SLM materials showed a microstructure free of PPB. The mechanical properties of specimens that underwent SLM + solution treatment and aging were comparable to those of conventional wrought specimens both at room temperature and 650 °C. However, a drop was observed in the ductility of HIP material at 650 °C. The brittle particles along the PPB were found to affect the HIP materials' creep life and ductility during solid-state sintering.

**Keywords:** superalloy; surface contamination; prior particle boundary; hot isostatic pressing; selective laser melting

# 1. Introduction

IN718 superalloy is widely used in gas turbine and related aerospace applications due to its excellent hot corrosion resistance, good weldability, and high mechanical properties—e.g., creep and fatigue strengths [1–3]. Macro segregation occurs during the conventional ingot metallurgy route due to melt-related problems such as segregation and porosity, which lead to degradation of the mechanical properties of the alloys [4,5]. One solution to this problem was to minimize segregation through rapid solidification of the metal or through the solid-state sintering process. A powder metallurgy technology-i.e., hot isostatic pressing (HIP)-was developed to eliminate segregation and macro-porosity, and to manufacture net shape components with homogeneous microstructure and improved mechanical properties [6]. Moreover, the temperature involved in the HIP process allows segregants to diffuse, thus enhancing the mechanical properties. After this type of solid-state sintering process, special care—i.e., rolling, extrusion and forging—is required to mitigate the detrimental effects of prior particle boundary (PPB) precipitation [7]. The so-called PPB problem results from surface contamination with pre-alloyed powder, and particles such as oxides, carbides, oxy-carbides and oxycarbonitrides are formed along the PPB [8–10]. These precipitates along the PPB are brittle and thus provide an easy fracture path, and so the undesirable effects of PPBs have been focused on because they limit the applications of such consolidated products in HIPed materials [11]. Although many efforts have been made to decrease the effects of PPBs to extend the life of such products in applications in the aerospace industry, no effective and reliable method has yet been determined. Hence, an additive manufacturing technology, selective laser melting (SLM), was used in the current research. The laser beam irradiates metal powders until they fully melt in the SLM process, which has the benefit of causing PPB breakdown. SLM is applicable to components of fully dense bulk material with complex



geometry [12]. In spite of the fact that the process is capable of making full density (~98–99%) parts, post-processes such as HIP were commonly used to eliminate the small amount of residual porosity for critical applications where high strength and fatigue resistance were necessary [13,14]. The obvious characteristics of the mechanical properties of SLM parts include anisotropy; the reasons for the special characteristics of IN718 built up by SLM—e.g., the high dislocation density, subgrain boundary and a row of interdendritic  $\delta$ -phase precipitates—have been explained in our previous studies from several perspectives [15,16]. However, research related to the surface contamination on powder in SLM materials is limited. Therefore, the current study focused on surface contamination and on the consolidation behaviour between SLM and conventional HIP. In order to gain an in-depth understanding of the surface contamination by the powder both in the SLM and HIP processes, the microstructure-mechanical relationship of IN718 nickel-based alloy fabricated by HIP and SLM processes were compared in the current study.

#### 2. Materials and Experimental Procedure

The prealloyed IN718 superalloy powder was prepared by vacuum melting and atomized into powders using the gas-atomizing method. The gas-atomized IN718 powders used for both the HIP process and the SLM process were typically spherical particles that were finer and had a broad particle size distribution below 60 µm (Figure 1), which may provide high packing density and reduce the percentage of voids [17,18]. The chemical composition of IN718 powder used in the HIP and SLM processes is shown in Table 1. The oxygen contents in the powders used for HIP and SLM were comparable. For the HIP process, the powders were entirely canned into SUS304 stainless steel capsules which was heated and evacuated to  $1.0 \times 10^{-3}$  Pa by an oil-diffusion pump. An optimal HIP condition which consisting of a soaking temperature of 1180 °C and a pressure of 175 MPa for 4 h was used [19]. On the other hand, a set of SLM process parameters (laser power: 400 W; scanning speed: 7.0 m/s; layer pitch: 20 µm; layer thickness: 40 µm; beam diameter: 100 µm; atmosphere: pure 99.9999% argon) was utilized to fabricate an IN 718 cube with the dimensions  $35 \times 35 \times 35$  mm<sup>3</sup>. The SLM cube was built using the alternating scan vector with a  $66.7^{\circ}$  rotation of the scanning direction. The scanning strategy of SLM process that were used are illustrated in Figure 2a. After the HIP and SLM processes, the specimens were subjected to solution treatment and aging (STA): a solution treatment at 980 °C for 1 h/air cooling (AC) to room temperature and a two-step aging treatment consisting of 718 °C for 8 h/furnace cooling to 621 °C and holding at 621 °C for 10 h before AC to room temperature. The heat histories of the specimens are shown in Figure 3a,b. Tensile test and creep test specimens were cut from the HIPed ingots and cubed using a spark cutter, and the gage length of each was  $19.6 \times 2.8 \times 3.0 \text{ mm}^3$ (Figure 2b). The tensile tests were conducted at room temperature (25 °C) and at 650 °C under a strain rate of  $4.25 \times 10^{-4}$  s<sup>-1</sup>. The creep test was conducted at 650 °C and 550 MPa. The microstructures were observed by a scanning electron microscope (HITACHI S-3700N, Tokyo, Japan), and transmission electron microscope (TEM, JEOL JEM-3200FS, Tokyo, Japan). Inverse pole figures (IPF) were calculated from the orientation measurements by electron backscatter diffraction (EBSD, Oxford Instruments, Oxfordshire, UK). X-ray diffraction (XRD, Rigaku Ultima IV, Tokyo, Japan) was used to determine the relative amounts of phases in the specimens.



Figure 1. Cont.



**Figure 1.** Secondary electron (SE) images of gas-atomized IN718 powders used for (**a**) hot isostatic pressing (HIP) and (**b**) selective laser melting (SLM); SE images of (**c**) particles along the prior particle boundary (PPB) in an as-HIPed specimen and (**d**) an as-built SLM specimen.

Table 1. Chemical composition of IN718 powder used in HIP and SLM (mass %).

IN718	Cr	Nb	Mo	Ti	Al	Со	Cu	С	Si, Mn	P, S	В	0	Fe	Ni
HIP	18.4	4.98	3.3	0.84	0.57	0.1	0.01	0.026	0.08	0.02	0.001	0.014	17.2	Bal.
SLM	19.6	5.05	2.85	1.10	0.46	0.03	0.05	0.04	0.04	0.0	0.002	0.019	Bal.	52.59



**Figure 2.** Illustrations of (**a**) scanning strategy of the SLM process and (**b**) dimensions of the mechanical test specimen (mm).



Figure 3. Heat history of the (a) HIP process and (b) SLM process.

### 3. Results

#### 3.1. Microstructures of as-HIPed Specimen and as-Built Specimen

Densities of HIP and SLM, C&W alloys were 8.19, 8.18, and 8.20 g/cm<sup>3</sup> respectively, as determined by the Archimedes principle. Investigation by X-ray diffraction (XRD) revealed the major peaks of  $\gamma$  matrix, while the oxide peaks such as Al<sub>2</sub>O<sub>3</sub> or TiO<sub>2</sub> could not be identified due to the limited volume fraction (Figure 4). Earlier studies showed comparable results which did not show the oxides peaks in the as-HIPed and as-built SLM specimens [19–21]. However, the clearly continuous precipitates were localized along the PPB in the as-HIPed specimen from the SEM observations (Figure 1c). The precipitates along the PPB in the as-HIPed specimen were mainly oxides (Figure 5a,c). Al, which easily reacts with oxygen, was found to be the major metallic element in an oxide layer of  $Al_2O_3$  (Figure 5c). By contrast, the as-built specimen built up by the SLM process showed a free-PPB microstructure (Figure 1d). Oxides and carbides induced by oxygen and carbon contamination of pre-alloyed powder were rarely found in the as-built SLM specimen (Figure 5b). EBSD measurements were used to establish the crystallographic orientation of the  $\gamma$  matrix for all of the specimens. The orientations were colored in terms of a [001] inverse pole figure coloring scheme as shown in the legend (Figure 6). IPF maps of the as-HIPed and as-built specimens are shown in Figure 6a, b by means of EBSD, respectively. The equiaxed grains with an average size of 8.8 µm were observed in the as-HIPed specimen (Figure 6a), whereas the columnar grains were observed in the as-built specimen built up by the SLM process (Figure 6b). These columnar grains were attributed to the epitaxial growth and heat flux during the solidification in the SLM process [22].

## 3.2. Influence of the Heat-Treatment Process

IPF maps of the HIP + STA specimen and SLM + STA specimen are shown in Figure 6c,d, respectively. The SLM + STA specimen had IPF images similar to those of the as-built specimen (Figure 6b,d). For the HIP materials, the equiaxed grains structure could be observed both before and after the heat-treatment process (Figure 6a,c). The fine equiaxial grains were observed in the HIP materials because of the presence of the PPB, which would impede grain growth [23]. Moreover, carbides along the PPB appeared to increase after heat treatment when compared with as-HIPed material (Figure 7). The previous oxides along the PPB promoted the subsequent precipitation of carbides, since specific oxides act as nuclei for the formation of MC-type carbides or  $M_{23}C_6$ -type carbides along the PPB (Figure 7). During the heat-treatment process, titanium and carbon within the powder would migrate to the specific oxides on the particle surfaces, resulting in the formation of stable titanium oxy-carbides along the PPB [24].



Figure 4. X-ray diffraction (XRD) patterns of as-HIPed and as-built SLM samples.



**Figure 5.** Transmission electron microscopy (TEM) bright field micrograph of (**a**) as-HIPed specimen and (**b**) as-built SLM specimen; Digital maps of energy dispersive spectroscopy (EDS) analysis showing distribution of Al, O, Ti and Nb elements within (**c**) as-HIPed specimen and (**d**) as-built SLM specimen.



**Figure 6.** Inverse pole figures (IPFs) of (**a**) as-HIPed; (**b**) as-built SLM; (**c**) HIP + solution treatment and aging (HIP + STA) and (**d**) SLM + STA specimens were analyzed using the orientation measurements by electron backscatter diffraction (EBSD).

## 3.3. Tensile Properties

Engineering stress-strain curves and the tensile properties at room temperature and 650 °C are shown in Figures 8 and 9, Tables 2 and 3, respectively. At room temperature, the as-built specimen of SLM material showed lower tensile properties than those of the as-HIPed material (Figure 8). After the STA heat-treatment process, the proof and tensile strengths of the SLM materials were improved significantly compared with the as-built SLM materials (Table 2 and Figure 8). The tensile properties in the heat-treated materials were comparable to those of the conventionally wrought specimen at room temperature, as shown in Table 2. The principal strengthening phase in IN718 was the  $\gamma''$  precipitate that promoted the strengths of heat-treated materials (Figure 10). The ductility of SLM + STA materials

at room temperature was related to the fine dimples present on the fracture surface (Figure 11b). It was also noted that a mixed fracture surface which consisted of fine dimples and a few particle boundary facets was observed in the HIP + STA material (Figure 11a). The strengths and ductility of the SLM + STA specimen were comparable to those of the conventionally wrought specimens, both at room temperature and 650 °C, as shown in Tables 2 and 3. However, the ductility of the HIP + STA specimen was much lower than those of the conventionally wrought specimen and the SLM + STA specimen at 650 °C (Table 3). The ductility of the HIP + STA specimen was only one-tenth of that of the cast and wrought specimen at 650 °C (Table 3). A fracture surface was observed along the grain boundaries covered with PPB in the HIP + STA specimen (Figure 11c), whereas, fracture surface specimens were observed along the dendrite structure in the SLM + STA (Figure 11d).



**Figure 7.** TEM bright field micrographs of (**a**) particles along the PPB; (**b**) growth behavior of oxy-carbides; (**c**) EDS digital maps showing the distribution of elements within the HIPed material after the heat-treatment process.



Figure 8. Engineering stress-strain curves of HIP specimen and SLM specimens at room temperature.



Figure 9. Engineering stress-strain curves of HIP specimen and SLM specimens at 650  $^\circ$ C.

 Table 2. Tensile properties at room temperature.

	$\sigma_{0.2}$ (MPa)	$\sigma_{ m T}$ (MPa)	ε <sub>f</sub> (%)
As-HIPed	833	1225	32.1
As-built SLM	677	1023	28.1
HIP + STA	1125	1473	21.2
SLM + STA	1271	1425	18.6
Wrought + STA	1216	1435	24.0

	$\sigma_{0.2}$ (MPa)	$\sigma_{\mathrm{T}}$ (MPa)	$\varepsilon_{\mathrm{f}}$ (%)
As-HIPed	844	978.0	6.8
As-built SLM	594	862.0	25.1
HIP + STA	1031	1093	0.9
SLM + STA	1042	1142	10.1
Wrought + STA	1028	1139	10.5

**Table 3.** Tensile properties at 650  $^\circ\text{C}.$ 



**Figure 10.** TEM bright field micrograph of strengthening phase precipitate within (**a**) HIP + STA and (**b**) SLM + STA specimens and (**c**) high magnification of area I in (**b**).



**Figure 11.** Fracture surfaces of (**a**) HIP + STA specimen and (**b**) SLM + STA specimen tested at room temperature; fracture surfaces of (**c**) HIP + STA specimen and (**d**) SLM + STA specimen tested at 650 °C.

#### 3.4. Creep Properties

The creep curves of heat-treated specimens at 650 °C and 550 MPa are shown in Figure 12. As for the SLM + STA specimen, the creep rupture life was approximately 140 h, while the HIP + STA specimen exhibited a rupture life of only 20 h. The fracture surface was observed along the grain boundaries covered with PPB in the HIP + STA specimen as shown in Figure 13a.



Figure 12. Creep curves of HIP + STA specimen and SLM + STA at 650 °C and 550 MPa.



**Figure 13.** Fracture surfaces of (**a**) HIP + STA specimen and (**d**) SLM + STA specimen tested at 650 °C and 550 MPa.

#### 4. Discussion

## 4.1. Grain Morphologies of SLM Materials

Equiaxed grains with an average size of 8.8  $\mu$ m were observed in the HIP specimens (Figure 6a), whereas columnar grains were observed in the SLM specimens (Figure 6b). The columnar grains were attributed to the epitaxial growth which resulted from the partial melting of the back of the previous layer to promote growth from the prior grains of the previous layer in the [001] direction [25]. In addition, fine and equiaxed grains with a grain size of less than 1  $\mu$ m were also observed in the SLM specimens (Figure 6b). Fine grains can be formed by the constitutional supercooling mechanism [26].

#### 4.2. The Formation of PPB in HIP Materials

The presence of elements with high oxygen affinities in superalloys leads to the production of highly stable oxides at the powder surface, or so-called PPB, in the consolidated products. The effects of PPBs are known to have detrimental effects on the mechanical properties of HIP materials. If the amount of oxygen is limited, the formation of  $Al_2O_3$  and  $TiO_2$  is expected to occur first, before the other elements are oxidized, due to the order of Gibbs energies for the oxides [27,28] (see Figure 14). These oxides would promote the subsequent precipitation of MC-type carbides or  $M_{23}C_6$ -type carbides along the PPB (Figure 7). These oxides act as heterogeneous nuclei sites, which have a strong effect on the nucleation and growth behavior of the carbide. During the heat-treatment process, carbide-forming elements within the powder would migrate to the specific oxides of particle surfaces, resulting in stable oxy-carbides on the PPB [24].



Figure 14. Illustration of equilibrium oxygen partial pressure for some metal oxides/metal systems, reproduced with permission from [28]. Elsevier, 2012.

#### 4.3. The Effects of PPB on the Mechanical Properties of HIP Specimens at 650 °C

The strengths and ductility of the SLM + STA specimen were comparable to those of the conventionally wrought specimen, both at room temperature and 650 °C, as shown in Tables 2 and 3. However, a drop was observed in the ductility of HIP + STA material at 650 °C (Table 3). The fracture surfaces appeared completely along the PPB (Figure 11c). The presence of a PPB decorated with brittle oxide and carbide particles resulted in premature failure [11,29]. The degree of ductility of a material is related to the relaxation process that occurs at the crack-tip. When brittle non-metallic particles which tend to lead to interface decohesion are dispersed along the boundaries of the matrix content [30], such as carbides or oxides along the PPB in HIP materials, the process of crack nucleation at the grain boundary takes place more easily at 650 °C. Due to the interface between incoherent particles, the matrix becomes a preferred site for crack initiation [23]. The crack would begin to propagate rapidly

when the stress reaches a critical value. The presence and specific location of the precipitates along the PPB weaken the metallic bonding between two adjacent powders, which leads to worse mechanical properties [31], contributing to the predominance of fractures along the PPB in HIP materials at 650 °C (Figure 11).

Interestingly, the HIP + STA material strengthened by precipitation showed only one-seventh of the ductility shown by the as-HIPed specimen at 650 °C (Table 3). The drop in the ductility of the HIP + STA material was observed under both the tensile test and creep test at 650 °C. As the temperature increases above 750 °C, the coherent relationship between  $\gamma$  and  $\gamma''$  could be lost and be replaced by the incoherent  $\delta$  phase with the same Ni<sub>3</sub>Nb composition [16]. The  $\gamma''$  phase will be metastable, and the conversion reaction of  $\gamma/\gamma'' \rightarrow \delta$  accelerates at temperatures above 750 °C [32]. These needle-shaped and brittle  $\delta$  phases are generally undesirable due to their adverse effects on mechanical properties [33], which lead to " $\delta$ -phase embrittlement" [16,34] (Figure 12). Furthermore, the previous oxides along the PPB would promote the subsequent precipitation of carbides along the PPB (Figure 7). During the heat-treatment process, carbide-forming elements within the powder would migrate to the specific oxides of particle surfaces, resulting in stable oxy-carbides on the PPB [24]. The carbides would grow during the heat-treatment process, which includes enough diffusion time for the required carbide-forming elements to be obtained (Figure 7a,b). These script-like carbides or needle-shaped  $\delta$  phases generally would have a negative effect on alloy performance. Since the oxides, carbides and needle-shaped  $\delta$  phases are closely spaced along the PPB (Figure 7a), they provide a continuous path for crack propagation. At temperatures above 980 °C, Al<sub>2</sub>O<sub>3</sub> forms as a protective oxide scale [35]. After one hour of isothermal oxidation in CM-247LC, continuous Al<sub>2</sub>O<sub>3</sub> formed at 1150 °C, whereas, NiO formed at 1000 °C [36]. This means that the HIP temperature of 1180 °C would be suitable for forming  $Al_2O_3$  in Ni-based superalloys. The main reason for this is that PPB precipitate networks formed during HIPing would be maintained at the HIP temperature for a long time (Figure 3a). As a result, the HIP + STA material would exhibit premature fracturing with poor ductility at 650 °C.

## 4.4. Free-PPB Microstructure in SLM Materials

Precipitates were localized along the PPB in both the as-HIPed and HIP + STA specimens (Figure 1), whereas SLM materials showed a free-PPB microstructure despite having a comparable oxygen content (Table 1). The difference in microstructure can be explained as follows. The HIP process is a solid-state diffusion process occurring below the material's melting temperature. The presence in superalloys of elements with high oxygen affinities brings about highly stable oxides at the powder surface. These oxides along the PPB would become diffusion barriers, resulting in a slower diffusion process [37]. As for the SLM process, the laser beam irradiates metallic powder in order to form a layer-by-layer material in a chamber with an inert gas atmosphere. The main reason that limited oxides are observed would be that the precipitates along the PPB precipitates network are churned and spread uniformly. Another layer of metallic powder is then fused and deposited, and the previous layer absorbs the extra power, resulting in the remelting of a part of the previous layer [26]. This remelting process is analogous to dilution, which provides a continuous solid-liquid interface free of contaminants or oxides and a sufficient bonding between layers [26,38,39]. Furthermore, the evaporation of Al and the ejection of pure-Al-rich particles with only 0.4 wt % oxygen were observed during the SLM process [40]. These results show that Al is easy to vaporize, and that  $Al_2O_3$  does not easily form during the SLM process because the time exposed to high temperatures above 1150 °C is limited (Figure 3b). From the view point of thermodynamics, Figure 14 shows the equilibrium oxygen partial pressure for some metal oxides/metal systems [28]. Figure 14 clearly indicates that every oxide could be avoided when the optimal temperature and oxygen partial pressure are reached. Under the conditions below each curve in the figure (low oxygen partial pressure and high temperature), the oxide is decomposed. On the other hand, above each curve (i.e., under high oxygen partial pressure and low temperature conditions) the metal is easily oxidized [28]. In the current study, since the temperature was maintained

in the HIP process (1180 °C for 4 h), a certain degree of oxidation cannot be avoided under normal HIP conditions. HIP was performed in a chamber with a vacuum which contained contaminants in the form of water and oxygen. As a result, metals could be easily oxidized during the HIP process (Figures 3a and 14). On the other hand, the minimum temperature in a high-energy laser-beam-focused zone ranges from 1260 to 1336 °C (melting point of Alloy718) during the SLM process [35]. As a result of the decomposition of contaminant compounds, such compounds might appear at low density in SLM materials (Figures 1 and 5).

# 5. Conclusions

The following conclusions can be drawn from this work:

- (1) Continuous precipitates were clearly localized along the PPB in HIP materials, while SLM materials showed a microstructure free of PPBs and an extremely low number of contaminants.
- (2) The columnar grains in the SLM specimens were attributed to epitaxial growth.
- (3) The strengths and ductilities of the SLM + STA specimen were comparable to those of the conventionally wrought specimen both at room temperature and 650 °C. However, a drop was observed in the ductility of the HIP material at 650 °C.
- (4) The brittle precipitates formed in the HIP materials, such as oxy-carbides and δ phases, tended to lead to interface decohesion and resulted in premature fracture with poor ductility at 650 °C.
- (5) The main reason that PPB precipitate networks formed during HIPing is that the HIP temperature was held at a temperature suitable for forming Al<sub>2</sub>O<sub>3</sub> for a long time.
- (6) In the SLM process, Al<sub>2</sub>O<sub>3</sub>, which became a nucleation site of precipitates, is not easily formed, and the PPB precipitates network is churned and spread uniformly during the process.

**Acknowledgments:** This research was supported by the ALCA Program of the Japan Science and Technology Agency, JST, and a Grant-in-Aid for Scientific Research (16K06799) from the Japan Society for the Promotion of Science. This support is gratefully acknowledged.

**Author Contributions:** Yen-Ling Kuo performed the experiments, analyzed the data and wrote the paper; Koji Kakehi designed the experiments and wrote the paper.

Conflicts of Interest: The authors declare no conflict of interest.

# References

- 1. Yeh, A.C.; Lu, K.W.; Kuo, C.M.; Bor, H.Y.; Wei, C.N. Effect of serrated grain boundaries on the creep property of Inconel 718 superalloy. *Mater. Sci. Eng. A* **2011**, 530, 525–529. [CrossRef]
- 2. Chamanfar, A.; Sarrat, L.; Jahazi, M.; Asadi, M.; Weck, A.; Koul, A.K. Microstructural characteristics of forged and heat treated Inconel-718 disks. *Mater. Des.* **2013**, *52*, 791–800. [CrossRef]
- 3. Ghosh, S.; Yadav, S.; Das, G. Study of standard heat treatment on mechanical properties of Inconel 718 using ball indentation technique. *Mater. Lett.* **2008**, *62*, 2619–2622. [CrossRef]
- 4. Bai, Q.; Lin, J.; Tian, G.; Dean, T.A.J. Review and analysis of powder prior boundary (PPB) formation in powder metallurgy processes for nickel-based super alloys. *J. Powder Metall. Min.* **2015**, *4*. [CrossRef]
- 5. Lee, S.C.; Chang, S.H.; Tang, T.P.; Ho, H.H.; Chen, J.K. Improvement in the microstructure and tensile properties of Inconel 718 superalloy by HIP treatment. *Mater. Trans.* **2006**, *47*, 2877–2881. [CrossRef]
- 6. Rao, G.A.; Kumar, M. High performance stainless steel via powder metallurgy hot isostatic pressing. *Mater. Sci. Technol.* **1997**, *13*, 1027–1031. [CrossRef]
- 7. Gessinger, G.H. Powder Metallurgy of Superalloys; Butteriworth & Co.: London, UK, 1984.
- 8. Maurer, G.E.; Castledine, W.; Schweizer, F.A.; Mancuso, S. Development of HIP consolidated P/M superalloys for conventional forging to gas turbine engine components. In Proceedings of the 8th International Symposium on Superalloys, Pittsburgh, PA, USA, 28 September–1 October 2014; pp. 645–652.
- 9. Furrer, D.; Fecht, H. Ni-based superalloys for turbine discs. JOM 1999, 51, 14–17. [CrossRef]
- 10. Rao, G.A.; Srinivas, M.; Sarma, D.S. Effect of solution treatment temperature on microstructure and mechanical properties of hot isostatically pressed superalloy Inconel\* 718. *Mater. Sci. Technol.* **2004**, *20*, 1161–1170. [CrossRef]

- Bai, Q.; Lin, J.; Jiang, J.; Dean, T.A.; Zou, J.; Tian, G. A study of direct forging process for powder superalloys. *Mater. Sci. Eng. A* 2015, 621, 68–75. [CrossRef]
- Haase, C.; Bültmann, J.; Hof, J.; Ziegler, S.; Bremen, S.; Hinke, C.; Schwedt, A.; Prahl, U.; Bleck, W. Exploiting process-related advantages of selective laser melting for the production of high-manganese steel. *Materials* 2017, *10*, 56. [CrossRef] [PubMed]
- 13. Ellyson, B.; Brochu, M.; Brochu, M. Characterization of bending vibration fatigue of SLM fabricated Ti-6Al-4V. *Int. J. Fatigue* **2011**, *19*, 389–395. [CrossRef]
- 14. AlMangour, B.; Grzesiakc, D.; Yang, J.M. Selective laser melting of TiB2/H13 steel nanocomposites: Influence of hot isostatic pressing post-treatment. *J. Mater. Process. Technol.* **2017**, 244, 344–353. [CrossRef]
- Kuo, Y.L.; Horikawa, S.; Kakehi, K. The effect of interdendritic δ phase on the mechanical properties of Alloy 718 built up by additive manufacturing. *Mater. Des.* 2017, 116, 411–418. [CrossRef]
- 16. Kuo, Y.L.; Horikawa, S.; Kakehi, K. Effects of build direction and heat treatment on creep properties of Ni-base superalloy built up by additive manufacturing. *Scr. Mater.* **2017**, *129*, 74–78. [CrossRef]
- 17. Donachie, M.J.; Donachie, S.J. *Superalloys: A Technical Guide*, 2nd ed.; ASM International: Geauga County, OH, USA, 2002.
- Zhang, J.; Singer, R.F. Hot tearing of nickel-based superalloys during directional solidification. *Acta Mater.* 2002, 50, 1869–1879. [CrossRef]
- 19. Chang, S.; Lee, S.; Tang, T.; Ho, H. Influences of soaking time in hot isostatic pressing on strength of inconel 718 superalloy. *Mater. Trans.* **2006**, *47*, 426–432. [CrossRef]
- 20. Jia, Q.; Gu, D. Selective laser melting additive manufacturing of Inconel 718 superalloy parts: Densification, microstructure and properties. *J. Alloys Compd.* **2014**, *585*, 713–721. [CrossRef]
- Zhang, B.; Lee, X.; Bai, J.; Guo, J.; Wang, P.; Sun, C.; Nai, M.; Qi, G.; Wei, J. Study of selective laser melting (SLM) Inconel 718 part surface improvement by electrochemical polishing. *Mater. Des.* 2017, 116, 531–537. [CrossRef]
- Kanagarajah, P.; Brenne, F.; Niendorf, T.; Maier, H.J. Inconel939 processed by selective laser melting: Effect of microstructure and temperature on the mechanical properties under static and cyclic loading. *Mater. Sci. Eng. A* 2013, *588*, 188–195. [CrossRef]
- Rao, G.A.; Kumar, M.; Srinivas, M.; Sarma, D.S. Effect of standard heat treatment on the microstructure and mechanical properties of hot isostatically pressed superalloy inconel 718. *Mater. Sci. Eng. A* 2003, *A355*, 114–125. [CrossRef]
- 24. Crompton, J.S.; Hertzberg, R.W. Analysis of second phase particles in a powder metallurgy HIP nickel-base superalloy. *J. Mater. Sci.* **1986**, *21*, 3445–3454. [CrossRef]
- Olakanmi, E.O.; Cochrane, R.F.; Dalgarno, K.W. A review on selective laser sintering/melting (SLS/SLM) of aluminium alloy powders: Processing, microstructure, and properties. *Prog. Mater. Sci.* 2015, 74, 401–477. [CrossRef]
- 26. DuPont, J.N.; Lippold, J.C.; Kiser, S.D. *Welding Metallurgy and Weldability of Nickel Base Alloys*; John Wiley & Sons, Inc.: Hoboken, NJ, USA, 2009.
- He, L.M.; Su, Y.-F.; Allard, L.F.; Lance, M.J.; Lee, W.Y. Effects of preoxidation on the nucleation and growth behavior of chemically vapor-deposited α-Al2O3 on a single-crystal Ni-based superalloy. *Metall. Trans. A* 2004, *35*, 1113–1124. [CrossRef]
- 28. Hryha, E.; Dudrova, E.; Nyborg, L. On-line control of processing atmospheres for proper sintering of oxidation-sensitive PM steels. *J. Mater. Process. Technol.* **2012**, *212*, 977–987. [CrossRef]
- 29. Wallace, W.; Wiebe, W.; Whelan, W.P.; Dainty, R.V.; Terada, T. The effect of grain-boundary structure on the tensile fracture behavior of hot-isostatically pressed 713 LC Alloy Compacts. *Powder Metall.* **1972**, *16*, 416–435. [CrossRef]
- Tetelman, A.S.; Mcevily, A.J., Jr. Fracture of Structural Materials; John Wiley & Sons Inc.: New York, NY, USA, 1967; pp. 193–233.
- 31. Broek, D. Elementary Engineering Fracture Mechanics; Martinus Nijhoff: Leiden, The Netherlands, 1984.
- Idell, Y.; Levine, L.E.; Allen, A.J.; Zhang, F.; Campbell, C.E.; Olson, G.B.; Gong, J.; Snyder, D.R.; Deutchman, H.Z. Unexpected δ-phase formation in additive-manufactured Ni-based superalloy. *JOM* 2016, *68*, 950–959. [CrossRef]
- 33. Campo, E.; Turco, C.; Catena, V. The correlation between heat treatment, structure and mechanical characteristics in Inconel 718. *Metall. Sci. Technol.* **1985**, *3*, 16–21.

- Radavich, J.F.; Korth, G.E.; Antolovich, S.D.; Stusrud, R.W.; MacKay, R.A.; Khan, T.; Kissinger, R.D.; Klarstrom, D.L. High temperature degradation of Alloy 718 after longtime exposures. In Proceedings of the 7th International Symposium on Superalloys, Seven Springs, PA, USA, 20–24 September 1992; pp. 497–506.
- 35. Davis, J.R. ASM Specialty Handbook: Heat-Resistant Materials; ASM International: Geauga County, OH, USA, 1997.
- Chiou, M.S.; Jian, S.R.; Yeh, A.C.; Kuo, C.M. Effects of aluminum addition on the high temperature oxidation behavior of CM-247LC Ni-based superalloy. *Int. J. Electrochem. Sci.* 2015, *10*, 5981–5993.
- 37. German, R. Powder Metallurgy and Particulate Materials Processing: The Processes, Materials, Products, Properties and Applications; Metal Powder Industries Federation: Princeton, NJ, USA, 2005.
- 38. Abe, F.; Osakada, K.; Shiomi, M.; Matsumoto, M.; Shiomi, M. The manufacturing of hard tools from metallic powders by selective laser melting. *J. Mater. Process. Technol.* **2001**, *111*, 210–213. [CrossRef]
- 39. Das, S.; Beaman, J.J. Direct Selective Laser Sintering of Metals. U.S. Patent 6,676,892 B2, 13 January 2004.
- Wang, X.; Read, N.; Carter, L.N.; Ward, R.M.; Attallah, M.M. Defect formation and mitigation in selective laser melting of high γ' Ni-base superalloys. In Proceedings of the 13th International Symposium on Superalloys, Seven Springs, PA, USA, 11–15 September 2016; pp. 351–358.



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