

Article



A Structure for Accurately Determining the Mass and Center of Gravity of Rigid Bodies

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Abstract: Measuring the mass and Center of Gravity (CG) of rigid bodies with a multi-point weighing method is widely used nowadays. Traditional methods usually include two parts with a certain location, i.e., a fixed platform and a mobile platform. In this paper, a novel structure is proposed to adjust the mobile platform for eliminating side forces which may load on the load cells. In addition, closed-form equations are formulated to evaluate the performance of the structure, and transformation matrices are used to estimate the characteristics of the structure. Simulation results demonstrate that repeatability of the proposed structure is higher than the traditional one and there are no side forces. Moreover, the measurement results show that the relative error of mass was within 0.05%, and the error of CG was within ±0.3 mm. The structure presented in this paper provides a foundation for practical applications.

Keywords: mass; center of gravity; side force; load cell; mechanical structure; modeling

1. Introduction

Mass and Center of Gravity (CG) are prerequisite parameters when designing the dynamic performance of an aerospace vehicle, car, etc. [1,2]. When the shape of the vehicle is complicated, it is necessary to obtain mass and CG by experiments [3]. The mass can be achieved by weighing machines, for instance, load cells. There are several methods to measure the CG, which can be roughly classified into three categories as follows: (1) static method such as multi-point weighing method or unbalance moment method [1,4,5], (2) dynamic method such as spin balance method or inverted torsion pendulum method [6–9], and (3) other methods such as the photogrammetric technique for determination of the CG for large scale objects [10].

Each of these methods has its own strengths and weaknesses. For instance, the multi-point weighing method, which uses three or more load cells to support a test platform, has been widely used, and the reason is that this method can measure the mass and CG simultaneously. This method only depends on gravity force acting through the CG, and CG location is calculated by static equilibrium from the reaction forces on the load cells [1,4]. Due to these reasons, it is the most suitable method for some massive objects [4], and it is also the cheapest automatic system.

However, there are shortcomings in the multi-point weighing method, for example, a load cell is a spring and has a designed loading axis. In addition, the generated force against the load cell outside of this axis can result in errors and may shorten the operating life of the load cell [11–14]. Besides, the measurement accuracy of CG is related to the position of the loading force.

There are several types of load cells or bears to avoid incorrect loading, such as self-centering pendulum load cell, pendulum bearing, and pendulum supports. These load cells automatically guide the superstructure back to its original position when the load is introduced with lateral

displacement [15]. However, these methods are usually used for a single load cell. Towards the multi-point weighing method where three or more load cells need to be coupled, no related literatures report the measurement structure.

In order to achieve high accuracy when measuring mass and CG, optimizing the mechanical structure to make the force load on the designed axis is necessary. This paper focuses on designing a structure used in a multi-point weighing method to avoid lateral forces and get high repeatability. Moreover, mathematical modeling of the structure is further established to analyze the proposed method. Finally, simulation and experiment results are conducted to validate the effectiveness of the proposed method.

2. Background and Related Work

2.1. Method for Obtaining the Mass and CG of a Body

As shown in Figure 1, the multi-point weighing method is a static method, which includes three load cells and a support platform. These three load cells are placed on the same horizontal flat and angularly spaced by 120° . This proposed method depends only on gravity force acting through the CG [1,4].



Figure 1. Diagram of the multi-point weighing method.

We record the readings of the three load cells, and the sum of the three load cell readings is the mass of the whole system. Similarly, we also record the mass of the system without the rigid body, and the mass of the rigid body is obtained from the difference of the readings shown in Equation (1).

$$M = \Delta m_1 + \Delta m_2 + \Delta m_3 \tag{1}$$

where: *M* is the mass of the rigid body, and Δm_1 , Δm_2 , Δm_3 are the difference readings of the three load cells.

Based on the static equilibrium (first law), the CG location can be obtained by Equation (2).

$$\begin{cases} x = \frac{x_1 \cdot \Delta m_1 + x_2 \cdot \Delta m_2 + x_3 \cdot \Delta m_3}{M} \\ y = \frac{y_1 \cdot \Delta m_1 + y_2 \cdot \Delta m_2 + y_3 \cdot \Delta m_3}{M} \end{cases}$$
(2)

According to Equation (1) and Equation (2), there are two factors affecting the accuracy of the results.

(a) Accuracy of the load cell.

The load cell is sensitive to side forces, bending and torsional moments, which should be avoided in the measurement systems [11–16]. Figure 2 shows some examples of correct loading as well as incorrect loading methods on a load cell [13–15].



Figure 2. Correct loading on a load cell and some examples of incorrect loading. (a) Central load application; (b) Non-central load application; (c) Non-axial load application or side forces; (d) Moment-loading (torsion) load application.

(b) Location of the reaction forces (F_1 , F_2 and F_3) of the load cells.

The load cell always has a designed axis of loading. If the location of the reaction force is out of this designed axis, it will cause errors of the load cells' readings, and errors of the CG results can be calculated according to Equation (2).

To avoid these issues, some efforts are done by designing the structure act on the load cell properly.

2.2. Traditional Structure and the Improved Struture

Figure 3 shows some types of structure. (a) Cone joint with conical pan, which is a traditional mounting and it is used for individual load cells, (b) and (c) are types of rocker bearings. The distortion of the load cell's reading is practically absent for rocker bearings, because in this case only slight rolling friction is present instead of a bending stress. However, the horizontal restraint of the rocker bearing is significantly less than fixed bearings. Rocker bearings are recommended if the position of the superstructure only changes horizontally [15].



Figure 3. The construction of the traditional structure. (**a**) Cone joint with conical pan; (**b**) Ball joint with groove; (**c**) Ball joint with fillet.

We have researched CG determination for many years, and one traditional method is placing three load cells on the same horizontal flat, as shown in Figure 4. In particular, each load cell has a ball groove and each groove holds a ball, making the center of the ball groove on the designed axis of the load cell. The upper platform has a similar construction. The ball can stay at the center of the ball groove because of the gravity.



Figure 4. The construction of the traditional structure.

However, due to manufacturing tolerance and imperfect assembly, axis 2 is not the same as axis 1 shown in Figure 5, making it hard to guarantee *F* pass through each designed axis of the load cell. Moreover, reducing these errors will increase the costs significantly.



Figure 5. Axes of the ball grooves.

Regarding the above issue, an improved structure is proposed, as shown in Figure 6. Similar to Figure 4, three load cells rest on the base platform, each of them has a ball groove and each groove holds a ball, ensuring the center of the ball groove goes through the designed axis of the load cell. The base platform should be horizontal and this structure should minimize the friction.



Figure 6. The construction of the new structure.

Specifically, the upper platform has one ball groove, one cylindrical groove and one planar surface. Because of the characteristic of the cylindrical groove and the plane, the upper platform can adjust its position and ensure F passes through the designed axis of the load cell without side force, even though these grooves may have position errors. A mathematical analysis of the structure will be performed in the next part.

3. Parametric Representation of the Structure

Here, for a convenient and simple explanation, the platform with the load cells is called the base-frame, and the upper platform is called the platform-frame [17]. The base-frame (O_B, X_B, Y_B, Z_B) and the platform-frame (O_P, X_P, Y_P, Z_P) are two right-handed orthonormal coordinate systems with (X_B, Y_B, Z_B) and (X_P, Y_P, Z_P) as their bases respectively, as shown in Figure 7. The three centers of ball grooves, which are held by load cells, always rest on the X_BY_B -plane. The origin O_B of the base-platform is the intersection of the bisectors of the three angles of the base frame triangle. Coordinates of the centers of the ball grooves.

The origin O_P of the platform-frame is the center of the ball groove shown in Figure 7, and the X_P axis is parallel with the axis of the cylindrical groove. The plane is parallel with the X_PY_P -plane. In the neutral configuration, the platform-frame has a general position and rotation with respect to the base-frame.



Figure 7. The schematic of the structure.

It is assumed that the external force acting on the platform-frame is known as well as the positions of the load cells, and it is assumed that B_i (i = 1, 2, 3) and P_i (i = 1, 2, 3) are unknowns. The significance of modeling is to analyze whether directions of $\overrightarrow{B_iP_i}$ (i = 1, 2, 3) are vertical to the X_BY_B-plane, which means that there are no side forces.

To analyze the structure, the coordinates of P_i (i = 1, 2, 3) or B_i (i = 1, 2, 3) should represent the same coordinate system (O_B, X_B, Y_B, Z_B), as shown in Figure 8, and P_i in base-frame (O_B, X_B, Y_B, Z_B) is shown in Equation (3):

$$[P_i]_B = T_{BP} + R_{BP} \cdot P_i \tag{3}$$

where $T_{BP} = [x, y, z]_B^T$ is the relative position vectors between $O_P X_P Y_P Z_P$ and $O_B X_B Y_B Z_B$. R_{BP} is the matrix representation of the rotation between $O_P X_P Y_P Z_P$ and $O_B X_B Y_B Z_B$. P_1 is the coordinate of the contact point between the ball and the ball groove; P_2 is the coordinate of the contact point between the ball and the cylindrical groove; and P_3 is the coordinate of the contact point between the ball and the plane in the platform-frame. $[P_i]_B$ is the coordinates of P_i (i = 1, 2, 3) expressed in $O_B X_B Y_B Z_B$.



Figure 8. The vector loop between ball and ball groove.

The rotation matrix R_{BP} is expressed by:

$$R_{BP} = \begin{bmatrix} q_{11} & q_{12} & q_{13} \\ q_{21} & q_{22} & q_{23} \\ q_{31} & q_{32} & q_{33} \end{bmatrix}$$
(4)

where:

$$q_{11} = \cos \gamma \cdot \cos \beta$$

$$q_{12} = \cos \gamma \cdot \sin \beta \cdot \sin \alpha - \sin \gamma \cdot \cos \alpha$$

$$q_{13} = \cos \gamma \cdot \sin \beta \cdot \cos \alpha + \sin \gamma \cdot \sin \alpha$$

$$q_{21} = \sin \gamma \cdot \cos \beta$$

$$q_{22} = \sin \gamma \cdot \sin \beta \cdot \sin \alpha + \cos \gamma \cdot \cos \alpha$$

$$q_{31} = -\sin \beta$$

$$q_{32} = \cos \beta \cdot \sin \alpha$$

$$q_{33} = \cos \beta \cdot \cos \alpha$$
(5)

 $\vec{l_i}$ is the direction of $\vec{B_iP_i}$ (*i* = 1, 2, 3), which can be described by Equation (6) [17]:

$$\vec{l_i} = [P_i]_B - B_i = (T_{BP} + R_{BP} \cdot P_i) - B_i$$
(6)

Ignore the effects of friction, the direction of $\vec{B_iP_i}$ always passes through the centers of the balls. As a result, the magnitudes of $\vec{l_i}$ are constant, which equal to the diameter of the balls, as shown in Equation (7):

$$\|l_i\| = D \tag{7}$$

 F_i are the forces loading on the load cells, and the direction of F_i expressed in O_B-X_BY_BZ_B can be expressed by the unit vector \vec{n}_{Bi} , as shown in Equation (8):

$$\vec{n}_{Bi} = \frac{\vec{l}_i}{\|\vec{l}_i\|} = \frac{1}{D} \cdot \vec{l}_i$$
(8)

While the direction of F_i expressed in O_P - $X_PY_PZ_P$ can be expressed by the unit vector $\vec{n_{Pi}}$, as shown in Equation (9):

$$\vec{n_{Bi}} = R_{BP} \cdot \vec{n_{Pi}} \tag{9}$$

The stability of the platform-frame must meet two conditions, namely, the force balance theorem and equilibrium of couples [18], which can be formulated in Equation (10):

$$\begin{bmatrix} \vec{n}_{p_1} & \vec{n}_{p_2} & \vec{n}_{p_3} \\ \vec{r}_1 \times \vec{n}_{p_1} & \vec{r}_2 \times \vec{n}_{p_2} & \vec{r}_3 \times \vec{n}_{p_3} \end{bmatrix} \begin{bmatrix} F_1 \\ F_2 \\ F_3 \end{bmatrix} = \begin{bmatrix} F \\ M \end{bmatrix}$$
(10)

where *F* and *M* are the external force and torque respectively, which act on the platform-frame as shown in Figure 9. Here, the $\vec{r_i}$ (*i* =1, 2, 3) are formulated as Equation (11):

$$\vec{r_i} = \vec{O_p P_i} = P_i(x_i, y_i, z_i)$$
(11)



Figure 9. External force diagram of the measuring system.

Ignore the effects of friction; the F_i always passes through the centers of the ball grooves so that it rests on the base-frame, so do the $\overrightarrow{n_{Bi}}$. As a result, the coordinates B_{0i} of the centers of the three balls grooves are calculated by Equation (12):

$$B_{oi} = B_i + R_B \cdot \vec{n}_{Bi} \tag{12}$$

where R_B are the radiuses of ball grooves, and B_i are the contact points between the balls and the ball grooves.

Similarly, the orientation of F_1 always passes through the center of the ball groove (P_{01x} , P_{01y} , P_{01z}) which lies on the platform-frame, making the center of the ball groove equal to the origin (0, 0, 0) of the O_P-X_PY_PZ_P, which can be formulated as Equation (13):

$$\begin{bmatrix} x_{P1} \\ y_{P1} \\ z_{P1} \end{bmatrix} - R_P \begin{bmatrix} n_{P1x} \\ n_{P1y} \\ n_{P1z} \end{bmatrix} = \begin{bmatrix} P_{01x} \\ P_{01y} \\ P_{01z} \end{bmatrix} = \begin{bmatrix} 0 \\ 0 \\ 0 \end{bmatrix}$$
(13)

where (x_{P1}, y_{P1}, z_{P1}) is the coordinate of P_1 , which is the contact point between the ball and the ball groove; R_P is the radius of the ball groove; $(n_{P1x}, n_{P1y}, n_{P1z})$ are the vectors of $\vec{n_{P1}}$.

The orientation of the $\overrightarrow{n_{P2}}$ (n_{P2x} , n_{P2y} , n_{P2z}) passes through the axis of the cylinder groove, because the axis of the platform (X_P) is parallel with the axis of the cylinder groove, as a result, we can get Equation (14):

$$\begin{bmatrix} x_{P2} \\ y_{P2} \\ z_{P2} \end{bmatrix} - R_P \begin{bmatrix} n_{p2x} \\ n_{p2y} \\ n_{p2z} \end{bmatrix} = \begin{bmatrix} x_{p2} \\ 0 \\ 0 \end{bmatrix}$$
(14)

where (x_{P2} , y_{P2} , z_{P2}) is the coordinate of P_2 , which is the contact point between the ball and the cylindrical groove; R_P is the radius of the cylindrical groove.

Besides, the normal vector of the plane is the orientation of $\vec{n_{P3}}$ (n_{P3x} , n_{P3y} , n_{P3z}) and is given by Equation (15):

$$\begin{bmatrix} 0\\0\\1 \end{bmatrix} = \begin{bmatrix} n_{p3x}\\n_{p3y}\\n_{p3z} \end{bmatrix}$$
(15)

The solution described here uses the geometric model to calculate the relative position and orientation between the base-frame and the platform-frame. Analysis of the unknowns from Equation (3) to Equation (15) will help to analyze the characteristics of the improved structure.

4. Simulation Results and Discussion

A simulation was carried out to demonstrate the feasibility of the proposed method. As illustrated in Figure 10, the solution employs a system of 27 equations. The corresponding unknown variables were solved iteratively by using a nonlinear numerical technique [19]. The inputs to the solver include B_{0i} , P_{01} , P_{02} , $\vec{n_{p3}}$, $||\vec{l_i}||$, F, and M. The outputs of the algorithm include B_i , P_i , F_i , T_{BP} and R_{BP} .

Known	Number of		I In Langeven Mariah Lang	Number of
Variables	Equations		Unknown variables	Equations
Boi (<i>i</i> =1, 2, 3)	9	System of 27		
Po1 (0, 0, 0)	3	Equations	<i>Bi</i> (<i>хві,уві,zві</i>) (<i>i</i> =1, 2, 3)	9
Po2 (XP2, 0, 0)	3	N	$P_i(x_{Pi}, y_{Pi}, z_{Pi})$ (<i>i</i> =1, 2, 3)	9
$\overrightarrow{n_{p3}} = (0,0,1)^{\mathrm{T}}$	3		F _i (<i>i</i> =1, 2, 3)	3
$\ \vec{l}_i\ _{(i=1,2,3)}$	3	Nonlinear	$T_{BP} = \begin{bmatrix} x, y, z \end{bmatrix}_{B}^{T}$	3
$F\left(x,y,z ight)$	3	Solver	$R_{BP} = \left[\alpha, \beta, \gamma\right]_{B}^{T}$	3
$M\left(x,y,z\right)$	3			

Figure 10. The solution for the proposed mathematical model.

With the proposed mathematical modeling, we analyze the traditional structure. Setting the offsets between axis 1 and axis 2, as shown in Figure 5, to 0 mm, 0.05 mm, 0.10 mm, ..., 1.00 mm, the side forces loading on the three load cells were obtained and shown in Figure 11. The offsets of the loading axes between the designed axes are listed in Figure 12. The inputs are listed in Appendix B.

From Figures 11 and 12, we can see that the traditional structure provides high accuracy without lateral forces in the ideal case. However, when taking the manufacturing or assembling errors into account, i.e., making the axis of the ball grooves on the base-frame shift the axis of ball grooves on the platform-frame, the results show that the offsets of the loading axes between the designed axes increased, so do the side forces.

The improved structure was also analyzed. With the offset or not, the side forces loading on the three load cells were always less than 0.01 N, and the offsets of the loading axes between the designed axes were less than 0.001 mm, which means that the accuracy of the improved method is higher than the traditional one.



Figure 11. The simulation results of the side forces.



Figure 12. The simulation result of offsets between the loading axes and the designed axes.

5. Physical Experiments

Mass and CG of a sample were measured 20 times with the traditional structure and the improved one respectively, as shown in Figure 13, and the standard deviations of the data are listed in Table 1, which means that the new structure has higher repeatability. The accuracy of the load cell is 0.02%, and the range of the load cell is (0–220) kg.



Figure 13. Experiment platform for measuring mass and CG.

	Deviation of the Mass (kg) $_$	Deviations of the CG (mm)	
		Along Axis of X	Along Axis of Y
The traditional structure	0.06	0.32	0.33
The improved structure	0.01	0.06	0.10

Table 1. The standard deviations of the mass and the CG.

Mass and CG of a large-sized standard sample was measured by a system with the new structure as shown in Figure 14. Major dimensions of the system are shown in Table 2, and the specification and design requirements of the system are shown in Table 3.



Figure 14. Experimental platform for measuring the mass and CG.

Table 2. Major dimensions of the system.

Item	Dimensions
Height of the device	1680 mm
Length of the device	2850 mm
Width of the device	2140 mm
Accuracy of the load cells	0.03% (F.S.)

Table 3. Specification and design requirements of the system.

Item	Dimensions	
Mass properties to be measured	Mass, CG along three orthogonal axes (X, Y, Z)	
Range of CG measurement	(0–3000) mm	
CG accuracy	±0.3 mm	
Range of mass measurement	(500–1000) kg	
Mass accuracy	±0.05%	

The mass and CG of this sample have been calibrated by a metrological authority. According to the calibration result, the mass is 779.43 kg, and the offset of CG from the centroid of the sample is less than 0.05 mm. The standard sample was placed at 20 different positions of the measurement system. The mass errors measured by the system are shown in Figure 15, and the CG errors are shown in Figure 16.



Figure 15. The measurement error of mass.



Figure 16. The measurement error of the CG.

According to the results of the mass measurement, the standard deviation of mass is 0.05 kg, and the maximum error is -0.07 kg. Meanwhile, the standard deviation of the CG along X axis is 0.04 mm, and the standard deviation of the CG along Y axis is 0.05 mm, which means that the improved structure can be used to measure the mass and CG with high accuracy.

6. Conclusions and Outlook

An improved structure for measuring the mass and CG was proposed in this paper. In addition, the analysis of the structure was presented, which includes the position and orientation of the platform-frame. Simulation and experimental results illustrate that the repeatability of the proposed structure is higher than the traditional one. In future work, the deflection deformation of the balls and the friction between the balls and the grooves will be considered, and the optimal radiuses of the grooves and the balls will be considered too.

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Conflicts of Interest: The authors declare no conflicts of interest.

Appendix A

Variable (<i>i</i> = 1, 2, 3)	Description	Variable (<i>i</i> = 1, 2, 3)	Description
$B_{0i}\left(x_{i},y_{i},z_{i}\right)$	The coordinates of the ball grooves in $O_B\mathchar`-X_B Y_B Z_B$	$T_{BP} = \left[x, y, z\right]_B^T$	The relative position vectors between O_P - $X_PY_PZ_P$ and O_R - $X_RY_RZ_R$
R_B	The radiuses of the ball grooves on base-frame	$R_{BP} = \left[\alpha, \beta, \gamma\right]_B^T$	The rotation between O_P - $X_PY_PZ_P$ and O_B - $X_BY_BZ_B$
O_B	The origin of O_B - $X_BY_BZ_B$	$\overrightarrow{r_i}$	The point of the force act upon platform-frame
O_B - $X_BY_BZ_B$	The coordinate system of the base-frame	F_i	The forces between balls and grooves
B_i	The contact points between the balls and the ball grooves	F(x, y, z)	External force
P ₀₁	The center of the ball groove on the platform-frame	$M\left(x,y,z ight)$	External torque
R_P	The radius of the ball groove on the platform-frame	D	The diameter of the balls
O_P	The origin of the platform-frame	$\overrightarrow{l_i}$	the direction of F_i
O_P - $X_PY_PZ_P$	The coordinate system of the platform-frame	$\overrightarrow{n_{Bi}}$	The unit vector of $\vec{l_i}$
P _i	The coordinates of the contact points between the balls and the grooves on the platform-frame	$\stackrel{\rightarrow}{n_{Pi}}$	$\vec{n_{Bi}}$ expressed in O _B -X _B Y _B Z _B
$[P_i]_B$	The coordinates of P_i expressed in O_B - $X_BY_BZ_B$		

Table A1. Variables and their description.

Appendix **B**

Table A2. Inputs and their values.

$B_{o1} = (400, 0, 0)^{\mathrm{T}}$	$P_{02} = (x_{P2}, 0, 0)^{\mathrm{T}}$
$B_{o2} = (-200, 346.41, 0)^{\mathrm{T}}$	$\overrightarrow{n_{p3}} = (0,0,1)^{\mathrm{T}}$
$B_{o3} = (-200, -346.41, 0)^{\mathrm{T}}$	$\ \vec{l_i}\ = 20$
$P_{01} = (0, 0, 0)^{\mathrm{T}}$	$F = (0, 0, 600)^{\mathrm{T}}$
$M = (0, 0, 0)^{\mathrm{T}}$	$R_{B} = 15$
$R_{P} = 15$	

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