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**Abstract:** This article focuses on designing an adaptive sliding mode controller via state and output feedback for nonlinear singular fractional-order systems (SFOSs) with mismatched uncertainties. Firstly, on the basis of extending the dimension of the SFOS, a new integral sliding mode surface is constructed. Through this special sliding surface, the sliding mode of the descriptor system does not contain a singular matrix *E*. Then, the sufficient conditions that ensure the stability of sliding mode motion are given by using linear matrix inequality. Finally, the control law based on an adaptive mechanism that is used to update the nonlinear terms is designed to ensure the SFOS satisfies the reaching condition. The applicability of the proposed method is illustrated by a practical example of a fractional-order circuit system and two numerical examples.

**Keywords:** sliding mode controller; singular fractional-order systems; state and output feedback; adaptive mechanism



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# 1. Introduction

Fractional order systems (FOSs) are especially suitable for describing the memory, heredity, mechanical and electrical properties of various materials, so they are widely used in various practical applications [1–3]. By constructing solvable linear matrix inequalities (LMIs), the robust control problems in FOSs can be effectively solved. In [4–6], the stabilization problem of FOSs is solved by designing an observer. In [7], an LMI based on criteria to ensure the stability or stabilization of FOSs for a given order  $\alpha$ :  $0 < \alpha < 2$  was provided.

The generalized system model can describe a broader practical system, which is more general than normal systems. Therefore, it is very important to study the generalized system model [8-10]. As for SFOSs, a large amount of research for the problems of admissibility and robust stabilization has been reported recently in [11]. In proving the admissibility of SFOSs, the authors in [12,13] solve the problem by transforming the SFOSs into normal FOSs through a specific feedback controller. On the other hand, when the fractional-order  $0 < \alpha < 1$ , the authors in [14] extend the admissibility method of integer order singular systems to SFOSs and present three different LMI conditions for the admissibility [15]. Correspondingly, in [16], this problem with  $1 < \alpha < 2$  is considered, whereas Theorem 3 in [16] is a bilinear matrix inequality, which is difficult to be used to solve the output feedback control problem of SFOSs. Since state variables are often difficult to obtain in practical applications, it is meaningful to research the design of the output feedback controller. The authors in [17] provide the LMI condition to solve the output feedback control problem by making the output matrix *C* satisfy a particular construct. The authors in [18] investigate the robust stabilization problem for T-S fuzzy FOSs via output feedback control, but this approach is difficult apply to SFOSs with  $0 < \alpha < 1$ . In [19], this defect has been overcome by using an intermediate variable to construct a strict LMI condition. However, this method needs to first know the information of the status variable, which is conservative.

As a mature control strategy, sliding mode control [20–23] (SMC) has recently received much attention in FOSs [24,25]. In practical applications, uncertain nonlinear systems [26] with disturbances are ubiquitous, and the SMC is a nonlinear robust control method, which can maintain the stability of systems. In [27], a reduced dimension of a sliding surface is constructed to get rid of the influence of nonlinear terms, and the control law is designed to force the FOSs to move along the sliding surface. A discrete-time fractional-order SMC scheme is proposed in [28]. In [29], the authors design a sliding mode controller for fractional-order financial systems. In [30], the integral sliding surface is designed for SFOSs, the attractive part of the this surface is that it enables the trajectory of SFOSs to start from the constructed sliding surface, which makes SFOSs more robust.

Any actual system has varying degrees of uncertainty. For example, the structure and parameters of the mathematical model describing the controlled object are not necessarily known by the designer in advance. In the face of these uncertainties, to design appropriate control function to make a specific performance index reach and maintain the optimum or approximate optimum, adaptive control [31] is proposed. Adaptive control, as a control method to estimate the nonlinear terms that are unknown in FOSs, is presented in [32]. In [33], the adaptive backstepping control schemes are proposed for FOSs. The authors in [34] used a fuzzy adaptive state observer to estimate the unmeasured state. In [35], the problem of adaptive control for uncertain T-S fuzzy FOSs with saturated control inputs is addressed. It is worth noting that adaptive control also plays an important role in the design of a sliding mode controller for FOSs [36-38]. The authors in [39] study the adaptive fuzzy SMC problem for chaotic FOSs. In [40], the authors propose the adaptive SMC method for nonlinear FOSs. For mismatched disturbances in nonlinear systems, the authors in [41] provide a novel scheme based on a disturbance observer to estimate the disturbances of FOSs. In [42], the problem of real-time SMC for dynamical systems is solved. Furthermore, the authors in [43] develop an adaptive sliding mode controller for time-varying delay singular systems. Furthermore, adaptive SMC is also of great significance in practical applications. For example, [44,45] studied adaptive SMC of rubber-tired gantry crane. By using adaptive fractional-order SMC, the controller can track the driving state well in the case of parameter uncertainty and unknown disturbance. Since the adaptive sliding mode controller can solve the matching uncertainty with an unknown upper bound and input saturation, it has more far-reaching significance. For example, it can eliminate the condition of the nonlinear term [46]. However, the design of the adaptive sliding mode controller via output feedback for SFOSs [47] with fractional-order  $0 < \alpha < 1$  is still a difficult and open problem.

Inspired by the aforementioned discussions, the main contributions of this paper are listed below.

(1) At present, most papers need feedback control for normal SFOSs [11–13]. Since the special sliding surface is constructed in this paper, which leads to the sliding motion being a normalized FOS, this approach can be regarded as a new normalization technique without the feedback control.

(2) On the basis of extending the dimension of the SFOS, a new integral sliding mode surface is constructed. Through this special sliding surface, the sliding mode of the descriptor system does not contain a singular matrix E. Thus, we can use the stability theorem of the normal system to study the stability of the sliding mode. In [8,9], their sliding modes contain a singular matrix E, which can only be studied by the admissibility theorem of the singular system.

(3) The SMC stabilization problems with state feedback and output feedback are both investigated, and the main results are in terms of LMIs. In addition, the output feedback SMC scheme does not need to calculate the intermediate variable, which is more effective than [19].

(4) The adaptive SMC law is proposed such that the finite-time reachability of the sliding surface can be guaranteed. Furthermore, it can be deal with the nonlinear terms.

The restriction that the norm of the disturbance of the system should be known in [4] is removed.

In the following, Section 2 describes the preliminaries. In Section 3, state and output feedback SMC is considered. In Section 4, the proposed methods are evaluated by a practical example and two numerical examples. Finally, conclusions of this study are provided in Section 5.

**Notations:** In this paper,  $M^T$  is the transpose of matrix M. Matrix X > 0 (< 0) means that the X is positive (negative) definite. \* indicates the symmetric part of a matrix, such as  $\begin{bmatrix} S & N \\ * & Z \end{bmatrix} = \begin{bmatrix} S & N \\ N^T & Z \end{bmatrix} \cdot || \cdot ||$  represents the induction norm of a matrix or the Euclidean norm of a vector.  $a = \sin(\frac{\alpha\pi}{2}), b = \cos(\frac{\alpha\pi}{2})$  and  $\operatorname{sym}(Y) = Y + Y^T$  for short in the sequel.

#### 2. Preliminaries

Consider the following nonlinear SFOS with fractional-order  $0 < \alpha < 1$ 

$$\begin{cases} E\mathcal{D}^{\alpha}x(t) = (A + \Delta A)x(t) + B(u(t) + g(x(t), t)),\\ y(t) = Cx(t), \end{cases}$$
(1)

where  $x(t) \in \mathbb{R}^n$ ,  $y(t) \in \mathbb{R}^p$  are the system state and the measured output, respectively.  $u(t) \in \mathbb{R}^l$  is the control input.  $E \in \mathbb{R}^{n \times n}$  is a singular matrix and rank(E) = m < n.  $A \in \mathbb{R}^{n \times n}$ ,  $B \in \mathbb{R}^{n \times l}$  and  $C \in \mathbb{R}^{p \times n}$ .  $\Delta A \in \mathbb{R}^{n \times n}$  is the mismatched uncertain matrix, assuming that  $\Delta A = UF(t)V$ , where U and V are appropriate dimensions matrices, and F(t) is an unknown matrix function which satisfies  $F^T(t)F(t) \leq I$ . Moreover, the nonlinear unknown function g(x(t), t) represents the matched uncertainty or disturbance, which is assumed to satisfy that

$$||g(x(t),t)|| \le \beta_1 + \beta_2 ||x(t)||, \tag{2}$$

where  $\beta_1$  and  $\beta_2$  are unknown positive real constants.

The Caputo fractional calculus of a function f(t) is defined as

$$\mathcal{D}^{\alpha}f(t) = \frac{1}{\Gamma(1-\alpha)}\int_{0}^{t}(t-\tau)^{-\alpha}f'(\tau)\mathrm{d}\tau,$$

where  $\Gamma(\cdot)$  is the Gamma function. It is easy to see that System (1) is equivalent to

$$\begin{cases} \widetilde{E}\mathcal{D}^{\alpha}\widetilde{x}(t) = \widetilde{A}_{\Delta}\widetilde{x}(t) + \widetilde{B}(u(t) + g(\widetilde{x}(t), t)), \\ y(t) = \widetilde{C}\widetilde{x}(t), \end{cases}$$
(3)

where

$$\widetilde{E} = \begin{bmatrix} I_n & 0 \\ 0 & 0 \end{bmatrix}, \quad \widetilde{A}_{\Delta} = \begin{bmatrix} 0 & I_n \\ A + \Delta A & -E \end{bmatrix}, \quad \widetilde{B} = \begin{bmatrix} 0 \\ B \end{bmatrix}, \\ \widetilde{x}(t) = \begin{bmatrix} x(t) \\ \mathcal{D}^{\alpha}x(t) \end{bmatrix}, \quad \widetilde{C} = \begin{bmatrix} C & 0 \end{bmatrix}, \quad g(\widetilde{x}(t), t) = g(x(t), t).$$

Here, related lemmas are introduced, which play a key role in dealing with SMC problems. We consider the unforced FOS as follows:

$$\mathcal{D}^{\alpha}x(t) = Ax(t). \tag{4}$$

**Lemma 1** ([14]). *System* (4) *is stable if and only if there exist matrices*  $X, Y \in \mathbb{R}^{n \times n}$ *, such that* 

$$\begin{bmatrix} X & Y \\ -Y & X \end{bmatrix} > 0, \tag{5}$$

$$sym(A(aX - bY)) < 0.$$
(6)

**Lemma 2.** If X is a positive definite matrix and Y is an antisymmetric matrix, then matrix aX - bY is nonsingular.

**Proof.** Assume that matrix aX - bY is singular, there exists a non-zero column vector  $\gamma$  that makes  $(aX - bY)\gamma = 0$ , so one obtains

$$\gamma^{\mathrm{T}}(aX - bY)\gamma = 0. \tag{7}$$

Since matrix *X* is positive definite and matrix *Y* is antisymmetric, Equation (8) is obtained by the transpose of Equation (7)

$$\gamma^{\mathrm{T}}(aX + bY)\gamma = 0. \tag{8}$$

By adding Equations (7) and (8), one has  $2a\gamma^{T}X\gamma = 0$ ; considering that *X* is a positive definite matrix, the assumption is wrong, so matrix aX - bY is nonsingular.

For SFOS output matrix C, there exists a singular value decomposition of C as follows

$$C = R_1 [Q \ 0] R_2^{\rm T}, \tag{9}$$

where matrix  $Q \in \mathbb{R}^{p \times p}$  is diagonal, and  $R_1 \in \mathbb{R}^{p \times p}$  and  $R_2 \in \mathbb{R}^{n \times n}$  are unitary matrices.

**Lemma 3** ([17]). There exists a matrix  $\hat{P}$  satisfying  $CP = \hat{P}C$  if and only if P is expressed as

$$P = R_2 \begin{bmatrix} X_{11} & 0 \\ X_{21} & X_{22} \end{bmatrix} R_2^{\mathrm{T}}.$$
 (10)

where  $X_{11} \in \mathbb{R}^{p \times p}$ ,  $X_{21} \in \mathbb{R}^{(n-p) \times p}$  and  $X_{22} \in \mathbb{R}^{(n-p) \times (n-p)}$ . The matrix  $\hat{P}$  is expressed as  $\hat{P} = R_1 Q X_{11} Q^{-1} R_1^{-1}$ .

## 3. Main Results

In this section, we study the adaptive sliding mode control of uncertain singular fractional-order systems by state feedback and output feedback, respectively.

#### 3.1. The Design of Sliding Mode State Feedback Controller

In this section, we design the fractional-order sliding mode state feedback controller. The design process includes, first, designing a novel sliding mode such that the sliding mode moves into normalized FOS. Second, we design an SMC so that the system can reach the above-described sliding surface in finite time.

We construct the following integral sliding surface function for System (3)

$$s(t) = G\widetilde{E}\mathcal{D}^{\alpha-1}\widetilde{x}(t) - \int_0^t G \ \widetilde{B}\widetilde{K}\widetilde{x}(\tau)d\tau,$$
(11)

where matrix  $G = [G_1 \ G_2]$ , and  $G_1, \ G_2 \in \mathbb{R}^{l \times n}$  are given matrices. We set that matrix G satisfies  $\det(G\widetilde{B}) \neq 0$ . It follows from  $G\widetilde{B} = G_2B$  that we can obtain  $\det(G_2B) \neq 0$ . Matrix  $\widetilde{K} = [K \ 0_{l \times n}]$  and  $K \in \mathbb{R}^{l \times n}$  need to be determined in the following part. When the SFOS moves on the sliding surface, one has  $\dot{s}(t) = 0$ . Thus, consider System (3), we have

$$\begin{split} \dot{s}(t) &= GA_{\Delta}\tilde{x}(t) + GBu(t) \\ &+ G\widetilde{B}g(\tilde{x}(t), t) - G\widetilde{B}\widetilde{K}\tilde{x}(t) = 0. \end{split}$$
(12)

Therefore, the equivalent control law is as follows

$$u_{eq}(t) = -(G\widetilde{B})^{-1}G(\widetilde{A}_{\Delta} - \widetilde{B}\widetilde{K})\widetilde{x}(t) - g(\widetilde{x}(t), t).$$
(13)

Equation (13) together with System (3) gives the sling motion in Equation (14):

$$\widetilde{E}\mathcal{D}^{\alpha}\widetilde{x}(t) = \widetilde{A}_{\Delta}\widetilde{x}(t) - \widetilde{B}(G\widetilde{B})^{-1}G\widetilde{A}_{\Delta}\widetilde{x}(t) + \widetilde{B}\widetilde{K}\widetilde{x}(t).$$
(14)

Letting  $\widetilde{G} = I_{2n} - \widetilde{B}(G\widetilde{B})^{-1}G$ , one has

$$\widetilde{G} = \begin{bmatrix} I_n & 0 \\ 0 & I_n \end{bmatrix} - \begin{bmatrix} 0 \\ B \end{bmatrix} (G_2 B)^{-1} \begin{bmatrix} G_1 & G_2 \end{bmatrix} = \begin{bmatrix} I_n & 0 \\ -B(G_2 B)^{-1}G_1 & I_n - B(G_2 B)^{-1}G_2 \end{bmatrix}.$$

We set  $\widetilde{G}_1 = -B(G_2B)^{-1}G_1$ ,  $\widetilde{G}_2 = I_n - B(G_2B)^{-1}G_2$ . Thus, Equation (14) is equivalent to

$$\begin{bmatrix} I_n & 0\\ 0 & 0 \end{bmatrix} \mathcal{D}^{\alpha} \widetilde{x}(t) = \begin{bmatrix} 0 & I_n\\ \widetilde{G}_2 A + \widetilde{G}_2 \Delta A + BK & \widetilde{G}_1 - \widetilde{G}_2 E \end{bmatrix} \widetilde{x}(t).$$
(15)

When  $G_1$  is a appropriate matrix,  $\tilde{G}_1 - \tilde{G}_2 E$  is invertible.

$$\mathcal{D}^{\alpha}x(t) = (\widehat{G}A + \widehat{G}\Delta A + \widehat{B}_1K)x(t), \tag{16}$$

where  $\widehat{G} = (\widetilde{G}_2 E - \widetilde{G}_1)^{-1} \widetilde{G}_2$  and  $\widehat{B}_1 = (\widetilde{G}_2 E - \widetilde{G}_1)^{-1} B$ .

**Remark 1.** By designing the sliding surface (Equation (11)), the sliding motion (Equation (16)) obtained is a normal system. This needs only one step. According to [13], it takes two steps to obtain the sliding motion of a normal system; the first step is to design the feedback control to normalize SFOSs, and the second step is to construct the sliding surface. Therefore, our method is superior to [13] in design.

**Theorem 1.** System (16) is stable if there exist matrices  $X, Y \in \mathbb{R}^{n \times n}$ ,  $Z \in \mathbb{R}^{l \times n}$  and a scalar  $\epsilon > 0$ , such that Equation (5) and the following LMI hold.

$$\begin{bmatrix} \operatorname{sym}(\widehat{G}A(aX-bY)+\widehat{B}_{1}Z)+\epsilon\widehat{G}UU^{\mathrm{T}}\widehat{G}^{\mathrm{T}} & (aX-bY)^{\mathrm{T}}V^{\mathrm{T}} \\ * & -\epsilon I \end{bmatrix} < 0, \quad (17)$$

then matrix K is designed as  $K = Z(aX - bY)^{-1}$ .

**Proof.** One can rewrite Equation (17) as the following inequality from the Schur complement.

$$sym((GA + B_1K)(aX - bY)) + \epsilon GUU^T G^T + \epsilon^{-1} (V(aX - bY))^T V(aX - bY) < 0.$$
(18)

According to Lemma 3 of [3], one has

$$sym((\widehat{G}A + \widehat{B}_1K)(aX - bY)) + sym(\widehat{G}\Delta A(aX - bY)) < 0.$$
(19)

Now, it is easy to see that System (16) is stable.  $\Box$ 

**Remark 2.** For singular systems, the advantage of the sliding motion being the normal system is that the gain matrix K can be calculated since it follows from Lemma 2 that matrix aX - bY is invertible. Compared with the sliding control scheme in [43], the sliding motion in [43] is a singular system, and the obtained gain matrix K needs to compute the inverse matrix of  $\tilde{E}W^T + Z^TS^T$  (see Theorem 4 in [43]). Since it involves an additional solved variable S that may cause the trouble that  $\tilde{E}W^T + Z^TS^T$  is not invertible.

In the following theorem, we design an adaptive SMC scheme to ensure the reachability of the slip mode. The adaptive parameters  $\hat{\beta}_1(t)$  and  $\hat{\beta}_2(t)$  are defined in order to estimate  $\beta_1$  and  $\beta_2$ , respectively.  $\bar{\beta}_1(t) = \hat{\beta}_1(t) - \beta_1$  and  $\bar{\beta}_2(t) = \hat{\beta}_2(t) - \beta_2$  represent the estimation errors. By the properties of fractional integrals, it yields as  $\dot{\bar{\beta}}_i(t) = \hat{\beta}_i(t)$ , where i = 1, 2.

**Theorem 2.** *Giving adaptive SMC law (20), System (3) can be driven into the sliding surface (Equation (11)) in finite time:* 

$$u(t) = (G\widetilde{B})^{-1} (G\widetilde{B}\widetilde{K}\widetilde{x}(t) - \lambda_0 s(t) - \rho(t) \operatorname{sgn}(s(t))),$$
(20)

where

$$\rho(t) = \lambda_1 + ||G\widetilde{B}||\widehat{\beta}_1(t) + ||G\widetilde{B}|||\widehat{\chi}(t)||\widehat{\beta}_2(t) + \sigma||G||||\widetilde{\chi}(t)||$$
(21)

with  $\lambda_0$  and  $\lambda_1$  being positive constants.  $\sigma$  is the norm bounded of the matrix  $\widetilde{A}_{\Delta}$ , which satisfies  $||\widetilde{A}_{\Delta}|| < \sigma$ . By increasing  $\lambda_0$ , the reaching time is shortened. By decreasing  $\lambda_1$ , the chattering is reduced. The above adaptive laws are chosen as

$$\widehat{\beta}_{1}(t) = \rho_{1}||s(t)|| ||G\widetilde{B}||, 
\widehat{\beta}_{2}(t) = \rho_{2}||s(t)|| ||\widetilde{x}(t)|| ||G\widetilde{B}||,$$
(22)

where  $\rho_1$  and  $\rho_2$  are designed as positive constants.

**Proof.** We choose the following Lyapunov function

$$V(t) = \frac{1}{2}s^{\mathrm{T}}(t)s(t) + \frac{1}{2\rho_{1}}\overline{\beta}_{1}^{2}(t) + \frac{1}{2\rho_{2}}\overline{\beta}_{2}^{2}(t).$$
(23)

By taking the derivative of V(t), we obtain

$$\dot{V}(t) = s^{\mathrm{T}}(t)\dot{s}(t) + \frac{1}{\rho_1}\overline{\beta}_1(t)\dot{\beta}_1(t) + \frac{1}{\rho_2}\overline{\beta}_2(t)\dot{\beta}_2(t).$$
(24)

With  $\dot{s}(t) = 0$  and Equations (20) and (24) are rewritten as

$$\dot{V}(t) = s^{\mathrm{T}}(t) (G\widetilde{A}_{\Delta}\widetilde{x}(t) - \lambda_0 s(t) - \rho(t) \mathrm{sgn}(s(t)) + G\widetilde{B}g(x(t), t)) + \frac{1}{\rho_1} \overline{\beta}_1(t) \dot{\widehat{\beta}}_1(t) + \frac{1}{\rho_2} \overline{\beta}_2(t) \dot{\widehat{\beta}}_2(t).$$
(25)

Thus, substituting Equation (21) into Equation (25), it follows that

$$\begin{split} \dot{V}(t) &\leq -\lambda_{0} ||s(t)||^{2} - \lambda_{1} ||s(t)|| - \beta_{1}(t)||GB|| \, ||s(t)|| \\ &- \widehat{\beta}_{2}(t)||s(t)|| \, ||x(t)|| \, ||G\widetilde{B}|| + \beta_{1}||G\widetilde{B}|| \, ||s(t)|| \\ &+ \beta_{2} ||s(t)|| \, ||x(t)|| \, ||G\widetilde{B}|| \\ &+ \frac{1}{\rho_{1}} \overline{\beta}_{1}(t) \widehat{\beta}_{1}(t) + \frac{1}{\rho_{2}} \overline{\beta}_{2}(t) \widehat{\beta}_{2}(t). \end{split}$$

$$(26)$$

Considering Equations (22) and (26), one has

$$\dot{V}(t) \le -\lambda_0 ||s(t)||^2 - \lambda_1 ||s(t)|| < 0, \ \forall ||s(t)|| \ne 0.$$
(27)

It is easy to see that System (3) moves to the sliding surface s(t) = 0 in finite time.  $\Box$ 

### 3.2. The Design of Sliding Mode Output Feedback Controller

In practical systems, the state of nonlinear system is difficult to obtain, so it is necessary to further investigate the sliding mode control problem based on output feedback. We give the following assumption first:

**Assumption 1.** The system uncertainty g(x(t), t) satisfies the following condition:

$$||g(\tilde{x}(t),t)|| \le \beta_3 + \beta_4 ||y(t)||,$$
(28)

*where*  $\beta_3$  *and*  $\beta_4$  *are unknown positive real constants.* 

*Then, we construct the following integral sliding surface function for System (3)* 

$$s(t) = S\widetilde{E}\mathcal{D}^{\alpha-1}\widetilde{x}(t) - S\int_0^t \widetilde{B}L\widetilde{C}\widetilde{x}(\tau)d\tau,$$
(29)

where matrix  $S = \begin{bmatrix} S_1 & S_2 \end{bmatrix}$ , and  $S_1, S_2 \in \mathbb{R}^{l \times n}$  are two given matrices. We set that matrix S satisfies  $det(S\widetilde{B}) \neq 0$  and rank  $\begin{bmatrix} S\widetilde{E} \\ \widetilde{C} \end{bmatrix} = rank \widetilde{C}$ . Matrix  $L \in \mathbb{R}^{l \times p}$  needs to be determined in the following part.

**Remark 3.** In order to ensure that there exists a matrix J that satisfies  $S\widetilde{E} = J\widetilde{C}$ , we propose a restriction condition that rank  $\begin{bmatrix} S\widetilde{E} \\ \widetilde{C} \end{bmatrix} = \operatorname{rank} \widetilde{C}$ . In this case, the surface function s(t) is converted to

$$s(t) = J\mathcal{D}^{\alpha-1}y(t) - S\int_0^t \widetilde{B}Ly(\tau)d\tau,$$
(30)

thus, the slide surface shown in Equation (30) only contains output information, which can be easily designed in practice.

From SMC theory, when the sliding motion takes place, it follows that s(t) = 0 and  $\dot{s}(t) = 0$ . From Equation (30),

$$\dot{s}(t) = S(\widetilde{A}_{\Delta} - \widetilde{B}L\widetilde{C})\widetilde{x}(t) + S\widetilde{B}u(t) + S\widetilde{B}g(\widetilde{x}(t), t) = 0.$$
(31)

Therefore, the equivalent control law is as follows

$$u_{eq}(t) = -(S\widetilde{B})^{-1}S(\widetilde{A}_{\Delta} - \widetilde{B}L\widetilde{C})\widetilde{x}(t) - g(\widetilde{x}(t), t).$$
(32)

By substituting Equation (32) into System (3), Equation (33) is obtained

$$\widetilde{E}\mathcal{D}^{\alpha}\widetilde{x}(t) = (\widetilde{A}_{\Delta} - \widetilde{B}(S\widetilde{B})^{-1}S\widetilde{A}_{\Delta} + \widetilde{B}L\widetilde{C})\widetilde{x}(t).$$
(33)

For notational simplicity, letting  $\tilde{S} = I_{2n} - \tilde{B}(S\tilde{B})^{-1}S$ , we have

$$\widetilde{S} = \begin{bmatrix} I_n & 0\\ 0 & I_n \end{bmatrix} - \begin{bmatrix} 0\\ B \end{bmatrix} (S_2 B)^{-1} \begin{bmatrix} S_1 & S_2 \end{bmatrix} = \begin{bmatrix} I_n & 0\\ -B(S_2 B)^{-1}S_1 & I_n - B(S_2 B)^{-1}S_2 \end{bmatrix}.$$

Defining  $\tilde{S}_1 = -B(S_2B)^{-1}S_1$  and  $\tilde{S}_2 = I_n - B(S_2B)^{-1}S_2$ , Equation (33) is represented as

$$\begin{bmatrix} I_n & 0\\ 0 & 0 \end{bmatrix} \mathcal{D}^{\alpha} \widetilde{x}(t) = \begin{bmatrix} 0 & I_n\\ \widetilde{S}_2 A + \widetilde{S}_2 \Delta A + BLC & \widetilde{S}_1 - \widetilde{S}_2 E \end{bmatrix} \widetilde{x}(t).$$
(34)

By choosing the appropriate matrix  $S_1$ , it follows that  $\tilde{S}_1 - \tilde{S}_2 E$  is invertible. Thus, Equation (34) is represented as

$$\mathcal{D}^{\alpha}x(t) = (\widehat{S}A + \widehat{S}\Delta A + \widehat{B}_{2}LC)x(t), \qquad (35)$$

where  $\widehat{S} = (\widetilde{S}_2 E - \widetilde{S}_1)^{-1} \widetilde{S}_2$  and  $\widehat{B}_2 = (\widetilde{S}_2 E - \widetilde{S}_1)^{-1} B$ .

**Theorem 3.** System (35) is stable if there exist matrices  $X_{11}$ ,  $Y_{11} \in \mathbb{R}^{p \times p}$ ,  $X_{12} \in \mathbb{R}^{p \times (n-p)}$ ,  $X_{22}$ ,  $Y_{22} \in \mathbb{R}^{(n-p) \times (n-p)}$ ,  $H \in \mathbb{R}^{l \times p}$  and a scalar  $\eta > 0$ , such that Equation (5) and the following LMI hold

$$\begin{bmatrix} \operatorname{sym}(\widehat{S}A(aX-bY)+\widehat{B}_{2}HC)+\eta\widehat{S}UU^{\mathrm{T}}\widehat{S}^{\mathrm{T}} & (aX-bY)^{\mathrm{T}}V^{\mathrm{T}} \\ * & -\eta I \end{bmatrix} < 0,$$
(36)

where

$$X = R_2 \begin{bmatrix} X_{11} & \frac{b}{a} X_{12} \\ \frac{b}{a} X_{12}^T & X_{22} \end{bmatrix} R_2^T, Y = R_2 \begin{bmatrix} Y_{11} & X_{12} \\ -X_{12}^T & Y_{22} \end{bmatrix} R_2^T,$$

 $R_1$ ,  $R_2$  and Q are defined in Equation (9), and then the matrix L is designed as  $L = HR_1Q(aX_{11} - bY_{11})^{-1}Q^{-1}R_1^{-1}$ .

**Proof.** One can rewrite Equation (36) as the following inequality from the Schur complement.

$$sym(\widehat{S}A(aX - bY) + \widehat{B}_2HC) + \eta \widehat{S}UU^{\mathsf{T}}\widehat{S}^{\mathsf{T}} + \eta^{-1}(V(aX - bY))^{\mathsf{T}}V(aX - bY) < 0.$$

$$(37)$$

According to Lemma 3 of [3], one has

1

$$\operatorname{sym}(\widehat{S}A(aX-bY)+\widehat{B}_2HC)+\operatorname{sym}(\widehat{S}\Delta A(aX-bY))<0. \tag{38}$$

According to  $H = LR_1Q(aX_{11} - bY_{11})Q^{-1}R_1^{-1}$  and using Lemma 3, one has

$$\widehat{B}_2 HC = \widehat{B}_2 LC(aX - bY). \tag{39}$$

Thus, Equation (40) is obtained

$$\operatorname{sym}((\widehat{S}A + \widehat{S}\Delta A + \widehat{B}_2 LC)(aX - bY)) < 0.$$
(40)

Combining to Lemma 1, System (34) is stable.  $\Box$ 

**Remark 4.** In [17], rankE = rank, C needs to be assumed first, and matrix C has a special structure  $C = [C_1 \ 0]$ , which is conservative. Theorem 3 in this paper does not require assumptions for the SFOSs, and the obtained result is more extensive.

**Remark 5.** Since the unknown matrices  $X_{11}$ ,  $X_{12}$ ,  $X_{22}$ , and the coefficient  $\frac{b}{a}$  are difficult to define X in the MATLAB LMI box, which results in the LMI condition in Equations (5) and (36) of the calculation Theorem 3. The next theorem overcomes this defect, which can be easily calculated in the MATLAB LMI box.

**Theorem 4.** System (35) is stable if there exist matrices  $\overline{X}_{11}$ ,  $\overline{Y}_{11} \in \mathbb{R}^{p \times p}$ ,  $\overline{X}_{12} \in \mathbb{R}^{p \times (n-p)}$ ,  $\overline{X}_{22}$ ,  $\overline{Y}_{22} \in \mathbb{R}^{(n-p) \times (n-p)}$ ,  $H \in \mathbb{R}^{l \times p}$  and a scalar  $\eta > 0$ , such that the following LMIs hold

$$\begin{bmatrix} \frac{\overline{X}}{a} & \frac{\overline{Y}}{b} \\ -\frac{\overline{Y}}{b} & \frac{\overline{X}}{a} \end{bmatrix} > 0,$$
(41)

$$\begin{bmatrix} \operatorname{sym}(\widehat{S}A(\overline{X}-\overline{Y})+\widehat{B}_{2}HC)+\eta\widehat{S}UU^{\mathsf{T}}\widehat{S}^{\mathsf{T}} & (\overline{X}-\overline{Y})^{\mathsf{T}}V^{\mathsf{T}} \\ * & -\eta I \end{bmatrix} < 0,$$
(42)

where

$$\overline{X} = R_2 \begin{bmatrix} \overline{X}_{11} & \overline{X}_{12} \\ \overline{X}_{12}^{\mathrm{T}} & \overline{X}_{22} \end{bmatrix} R_2^{\mathrm{T}},$$
$$\overline{Y} = R_2 \begin{bmatrix} \overline{Y}_{11} & \overline{X}_{12} \\ -\overline{X}_{12}^{\mathrm{T}} & \overline{Y}_{22} \end{bmatrix} R_2^{\mathrm{T}}$$

 $R_2$  has the same definition as in Theorem 3. Then, matrix L is designed as  $L = HR_1Q(\overline{X}_{11} - \overline{Y}_{11})^{-1}Q^{-1}R_1^{-1}$ .

**Proof.** There exist  $\overline{X}_{11} = aX_{11}$ ,  $\overline{X}_{12} = bX_{12}$ ,  $\overline{X}_{22} = aX_{22}$ ,  $\overline{Y}_{11} = bY_{11}$ , and  $\overline{Y}_{22} = bY_{22}$ ; then, we have  $\overline{X} = aX$  and  $\overline{Y} = bY$ . Equations (41) and (5) are equivalent to Equations (42) and (36), respectively.  $\Box$ 

The next theorem is proposed such that the reachability condition is ensured. Here, the adaptive parameters  $\hat{\beta}_3$ ,  $\hat{\beta}_4$  are defined to estimate  $\beta_3(t)$  and  $\beta_4(t)$ , respectively.  $\overline{\beta}_3(t) = \hat{\beta}_3(t) - \beta_3$  and  $\overline{\beta}_4(t) = \hat{\beta}_4(t) - \beta_4$  represent the estimation errors. When we obtain matrix *L* by LMIs (41) and (42), it is easy to see that the system state x(t) is bounded, and we set

$$\sup_{0 \le t < \infty} ||\widetilde{x}(t)|| \le \beta_5, \tag{43}$$

where  $\beta_5$  is an unknown positive constant. The adaptive parameter  $\hat{\beta}_5(t)$  is defined to estimate  $\beta_5$ , and the estimation errors are expressed as  $\bar{\beta}_5(t) = \hat{\beta}_5(t) - \beta_5$ . By the properties of fractional integrals, one obtains that  $\dot{\bar{\beta}}_i(t) = \hat{\beta}_i(t)$ , where i = 3, 4, 5.

**Theorem 5.** *Given the adaptive SMC law, System* (3) *is converged to the sliding surface* (Equation (29)) *in finite time.*  $\sim 1 - 2 - 2$ 

$$u(t) = (S\widetilde{B})^{-1}(S\widetilde{B}L\widetilde{C}\widetilde{x}(t) - \gamma_0 s(t) - \omega(t)\mathrm{sgn}(s(t))), \tag{44}$$

where

$$\omega(t) = \gamma_1 + ||S\widetilde{B}||\widehat{\beta}_3(t) + ||S\widetilde{B}|| ||y(t)||\widehat{\beta}_4(t) + \sigma||S||\widehat{\beta}_5(t)$$
(45)

with  $\gamma_0$  and  $\gamma_1$  being positive constants. By increasing  $\gamma_0$ , the reaching time is shortened. By decreasing  $\gamma_1$ , the chattering is reduced.  $\sigma$  is defined in Theorem 2. The above adaptive laws are chosen as

$$\begin{aligned}
\ddot{\beta}_{3}(t) &= \rho_{3}||s(t)|| ||SB||, \\
\dot{\hat{\beta}}_{4}(t) &= \rho_{4}||s(t)|| ||y(t)|| ||S\widetilde{B}||, \\
\dot{\hat{\beta}}_{5}(t) &= \rho_{5}\sigma||s(t)|| ||S||,
\end{aligned}$$
(46)

where  $\rho_3$ ,  $\rho_4$  and  $\rho_5$  are designed as positive constants.

Proof. We choose the following Lyapunov function

$$V(t) = \frac{1}{2}s^{\mathrm{T}}(t)s(t) + \frac{1}{2\rho_{3}}\overline{\beta}_{3}^{2}(t) + \frac{1}{2\rho_{4}}\overline{\beta}_{4}^{2}(t) + \frac{1}{2\rho_{5}}\overline{\beta}_{5}^{2}(t).$$
(47)

Therefore, the derivative of V(t) can be formulated as

$$\dot{V}(t) = s^{\mathrm{T}}(t)\dot{s}(t) + \frac{1}{\rho_{3}}\overline{\beta}_{3}(t)\dot{\widehat{\beta}}_{3}(t) + \frac{1}{\rho_{4}}\overline{\beta}_{4}(t)\dot{\widehat{\beta}}_{4}(t) + \frac{1}{\rho_{5}}\overline{\beta}_{5}(t)\dot{\widehat{\beta}}_{5}(t).$$

$$(48)$$

According to Equations (31) and (44), Equation (49) is derived as

$$\dot{V}(t) = s^{\mathrm{T}}(t)(S\widetilde{A}_{\Delta}\widetilde{x}(t) - \gamma_{0}s(t) - \varpi(t)\frac{s(t)}{||s(t)||} + S\widetilde{B}g(\widetilde{x}(t), t)) + \frac{1}{\rho_{3}}\overline{\beta}_{3}(t)\widehat{\beta}_{3}(t) + \frac{1}{\rho_{4}}\overline{\beta}_{4}(t)\widehat{\beta}_{4}(t) + \frac{1}{\rho_{5}}\overline{\beta}_{5}(t)\widehat{\beta}_{5}(t).$$

$$(49)$$

Thus, substituting Equation (45) into Equation (49), Equation (50) is obtained

$$\begin{split} \dot{V}(t) \leq & \beta_{5}||s(t)||\sigma||S|| - \gamma_{0}||s(t)||^{2} - \gamma_{1}||s(t)|| \\ & -\hat{\beta}_{3}(t)||s(t)|| ||S\widetilde{B}|| - \hat{\beta}_{4}(t)||s(t)|| ||y(t)|| ||S\widetilde{B}|| \\ & -\hat{\beta}_{5}(t)||s(t)||\sigma||S|| + \beta_{3}||s(t)|| ||S\widetilde{B}|| \\ & + \beta_{4}||s(t)|| ||y(t)|| ||S\widetilde{B}|| + \frac{1}{\rho_{3}}\overline{\beta}_{3}(t)\dot{\beta}_{3}(t) \\ & + \frac{1}{\rho_{4}}\overline{\beta}_{4}(t)\dot{\beta}_{4}(t) + \frac{1}{\rho_{5}}\overline{\beta}_{5}(t)\dot{\beta}_{5}(t). \end{split}$$
(50)

Considering Equations (46) and (50), one has

$$\dot{V}(t) \le -\gamma_0 ||s(t)||^2 - \gamma_1 ||s(t)|| < 0, \ \forall ||s(t)|| \ne 0.$$
(51)

It is easy to see that System (3) moves to the sliding surface (Equation (29)) in finite time.  $\Box$ 

**Remark 6.** When the derivative matrix E is nonsingular, the SMC scheme proposed in this paper is still valid for normal fractional-order systems. In addition, when the fractional-order  $\alpha = 1$ , the SMC scheme is also valid for integer-order systems.

**Remark 7.** In order to solve the problem that the system reaches the sliding surface in finite time, similar to the sliding mode control method of the integer-order system, we take the integer-order derivative of Lyapunov functional candidate V(t) in Theorems 2 and 5. Therefore, the problem of stability in finite time is solved.

### 4. Simulation Examples

In this section, we will use a practical example, a comparison with other article and a numerical example to illustrate the effectiveness of our results.

**Example 1.** We consider the fractional singular electrical circuit in [2] with given resistances  $R_i$ , i = 1, 2, where  $R_2$  is the nonlinear resistance with the voltage  $u_{R_2} = f(i_2)$ , inductances  $L_i$ , i = 1, 2, and source current  $i_z$ . The circuit is shown in Figure 1.



Figure 1. Electronic network.

Using Kirchhoff's laws, it is easy to see that

$$L_1 \frac{d^{\alpha} i_1}{dt^{\alpha}} + R_1 i_1 = L_2 \frac{d^{\alpha} i_2}{dt^{\alpha}} + R_2 i_2,$$
  
$$i_2 = i_1 + i_2$$

Thus, System (52) is obtained

$$E\mathcal{D}^{\alpha}x(t) = Ax(t) + Bu(t), \tag{52}$$

where

$$\begin{aligned} & \alpha = 0.4, \quad x(t) = \begin{bmatrix} i_1 & i_2 \end{bmatrix}^{\mathrm{T}}, \quad u(t) = i_z, \\ & E = \begin{bmatrix} L_1 & -L_2 \\ 0 & 0 \end{bmatrix}, \quad A = \begin{bmatrix} -R_1 & R_2 \\ -1 & -1 \end{bmatrix}, \\ & B = \begin{bmatrix} 0 \\ 1 \end{bmatrix}. \end{aligned}$$

When there are some disturbances in the circuit loop, that is, there are some uncertainties and disturbances, then System (52) can be written as follows.

$$E\mathcal{D}^{\alpha}x(t) = (A + \Delta A)x(t) + B(u(t) + g(x(t), t)).$$
(53)

We set  $L_i = 1$ , i = 1, 2,  $R_1 = 1$ ,  $R_2 = 1$ , and  $G_1 = \begin{bmatrix} 1 & 3 \end{bmatrix}$ ,  $G_2 = \begin{bmatrix} 0 & 1 \end{bmatrix}$ ,  $U = \begin{bmatrix} -1 & 1 \end{bmatrix}^T$ ,  $V = \begin{bmatrix} -1 & -2 \end{bmatrix}$ ,  $g(x(t), t) = x_2 \sin(x_2(t))$ . By using the LMI toolbox, Equations (5) and (17) are feasible, which indicates that the sliding motion of System (53) is stable, and the feasible solution is

$$X = \begin{bmatrix} 36.7641 & -12.2149 \\ -12.2149 & 21.1580 \end{bmatrix}, \quad Y = \begin{bmatrix} 0 & 0.4900 \\ -0.4900 & 0 \end{bmatrix}$$
$$Z = \begin{bmatrix} -24.1874 & -54.3561 \end{bmatrix}, \quad \epsilon = 34.9970,$$

matrix *K* is computed as

$$K = \begin{bmatrix} -3.0804 & -6.2473 \end{bmatrix}.$$

Then, through the designed adaptive SMC law (20), System (53) moves to the sliding surface in finite time.

In addition, we select

$$x(0) = [0.05 - 0.06]^{-1}$$

According to the above parameters, we obtain the simulation results. State responses of System (3) are drawn in Figure 2.



Figure 2. State trajectories of System (3) under the adaptive SMC law.

**Example 2.** Here, we compare our method with Theorem 4 of [43]. We consider uncertain SFOSs in System (3) with  $\alpha = 0.6$  and

$$E = \begin{bmatrix} 1 & 0 & 0 \\ 1 & 1 & -1 \\ 0 & 0 & 0 \end{bmatrix}, \quad A = \begin{bmatrix} -1 & 0 & -1 \\ 0 & 0 & 0 \\ 0 & -1 & -1 \end{bmatrix}, \quad B = \begin{bmatrix} 1 \\ 0 \\ 1 \end{bmatrix},$$
$$U = \begin{bmatrix} 1 & 0 & 0 \end{bmatrix}^{\mathrm{T}}, \quad V = \begin{bmatrix} 2 & 4 & 6 \end{bmatrix}.$$

The state feedback SMC problem is considered, the system uncertainty  $g(x(t), t) = x_1 \sin(x_1(t))$ ,  $G_1 = \begin{bmatrix} 1 & 2 & 5 \end{bmatrix}$  and  $G_2 = \begin{bmatrix} 1 & 0 & 0 \end{bmatrix}$  such that  $(G_2B)^{-1} = 1$ . Now, according to Theorem 1 and using the MATLAB LMI toolbox, we can obtain:

$$\begin{split} X &= \begin{bmatrix} 620.8555 & -116.5513 & -93.7812 \\ -116.5513 & 496.5470 & -225.0537 \\ -93.7812 & -225.0537 & 283.6576 \end{bmatrix}, \\ Y &= \begin{bmatrix} 0 & -3.4474 & 1.7696 \\ 3.4474 & 0 & -0.9207 \\ -1.7696 & 0.9207 & 0 \end{bmatrix}, \\ Z &= 10^3 \times \begin{bmatrix} -1.6802 & -0.9611 & -2.6413 \end{bmatrix}, \quad \epsilon = 786.5716. \end{split}$$

and

$$K = \begin{bmatrix} 11.7937 & 43.4645 & -58.4639 \end{bmatrix}.$$

Thus, System (16) is stable. By SMC law (20), System (3)'s trajectory can be driven to the sliding surface (Equation (11)) within finite time.

In addition, we select

$$x(0) = \begin{bmatrix} 1.3 & -0.02 & -1.1 \end{bmatrix}^{\mathrm{T}}$$

and initial estimates are given as  $\hat{\beta}_1(0) = 0.5$ ,  $\hat{\beta}_2(0) = 0.2$ . According to the above parameters, we used our method obtain the simulation results. State responses of System (3) are drawn in Figure 3. It reveals that its state x(t) converges to zero as  $t \to +\infty$ , and the resulting system is asymptotically stable. Figure 4 denotes the integral-type sliding mode surfaces s(t). Figure 5 plots the control input u(t). The adaptive parameters are shown in Figures 6 and 7.

For comparison, we give the followiing parameters to solve Theorem 4 in [43].

$$E_{0} = \begin{bmatrix} 1 & 0 & 0 \\ 1 & 1 & -1 \\ 0 & 0 & 0 \end{bmatrix}, A_{0} = \begin{bmatrix} -1 & 0 & -1 \\ 0 & 0 & 0 \\ 0 & -1 & -1 \end{bmatrix}, A_{d} = \begin{bmatrix} 3 & 5 & 5 \\ -1.5 & 1 & 1 \\ 3 & -1 & 10 \end{bmatrix} B_{0} = \begin{bmatrix} 1 \\ 0 \\ 1 \end{bmatrix},$$
$$B_{w} = \begin{bmatrix} 1 \\ 1 \\ 1 \end{bmatrix}, C_{0} = \begin{bmatrix} 1.5 \\ 2 \\ 2.5 \end{bmatrix}, C_{d} = \begin{bmatrix} 0.3 \\ 0.4 \\ 0.5 \end{bmatrix}, U_{1} = \begin{bmatrix} 0 & 0 & 0 & 1 & 0 & 0 \end{bmatrix}^{\mathrm{T}},$$
$$V_{1} = \begin{bmatrix} 2 & 4 & 6 & -0.1 & -0.1 & -0.1 \end{bmatrix}, V_{2} = \begin{bmatrix} -0.1 & 0.1 & 0.1 & 0 & 0 & 0 \end{bmatrix},$$
$$G = \begin{bmatrix} 1 & 2 & 5 & 1 & 0 & 0 \end{bmatrix}, D_{w} = 0.2, d_{0} = 0.2.$$

According to Theorem 4 in [43] solving LMI (45) of [43], we obtain

$$t = 0.0570.$$

From the result, we could not establish feasibility nor infeasibility. Therefore, the LMI is not strictly feasible.

From the above simulation, we can see that since aX - bY in our method must be invertible, the LMI of our method must have a feasible solution, but  $\tilde{E}W^{T} + Z^{T}S^{T}$  of Theorem 4 in [43] is not necessarily invertible. For example, the LMI in Theorem 4 has no strict feasible solution. We can use the simple MATLAB commands and figures in the Appendix A to show the uncertainty of matrix A and fractional-order model in Example 2.



Figure 3. State trajectories of System (3) under the adaptive SMC law.



**Figure 4.** Surface function s(t).



**Figure 5.** Adaptive SMC law u(t).



**Figure 6.** Adaptive parameter  $\hat{\beta}_1(t)$ .



**Figure 7.** Adaptive parameter  $\hat{\beta}_2(t)$ .

and

**Example 3.** Considering uncertain SFOSs in System (3).

ľ

U

$$\alpha = 0.6,$$

$$C = \begin{bmatrix} 0 & 1 & 0 \\ 1 & 0 & 0 \end{bmatrix}, E = \begin{bmatrix} 1 & 1 & 0 \\ 0 & 1 & 0 \\ 0 & 0 & 0 \end{bmatrix},$$

$$A = \begin{bmatrix} 3 & 5 & 5 \\ -1.5 & 1 & 1 \\ 3 & -1 & 10 \end{bmatrix}, \quad B = \begin{bmatrix} 1 \\ 1 \\ 1 \end{bmatrix},$$

$$= \begin{bmatrix} 1 & -1 & 2 \end{bmatrix}^{\mathrm{T}}, \quad V = \begin{bmatrix} 1 & 0 & -2 \end{bmatrix}.$$

The system uncertainly g(x(t), t) is assumed as  $g(x(t), t) = \sin(x_2(t))$ .  $S_1$  and  $S_2$  are chosen as  $S_1 = \begin{bmatrix} 1 & 1 & 0 \end{bmatrix}$  and  $S_2 = \begin{bmatrix} 3 & -1 & -1 \end{bmatrix}$ , respectively. It follows from Equation (9) that

$$R_{1} = \begin{bmatrix} -1 & 0 \\ 0 & -1 \end{bmatrix}, \quad Q = \begin{bmatrix} 1 & 0 \\ 0 & 1 \end{bmatrix},$$
$$R_{2} = \begin{bmatrix} 0 & -1 & 0 \\ -1 & 0 & 0 \\ 0 & 0 & 1 \end{bmatrix}.$$

It is found that LMIs (41) and (42) are feasible, and the solutions are

$$\overline{X} = \begin{bmatrix} 21.7043 & -19.1222 & -3.0762 \\ -19.1222 & 54.7421 & 4.9304 \\ -3.0762 & 4.9304 & 35.2072 \end{bmatrix},$$
  
$$\overline{Y} = \begin{bmatrix} 0 & -18.8349 & -3.0762 \\ 18.8349 & 0 & 4.9304 \\ 3.0762 & -4.9304 & 0 \end{bmatrix},$$
  
$$\eta = 2.6388, \ H = \begin{bmatrix} 159.9039 & -208.8719 \end{bmatrix}.$$
  
follows form  $L = HR_1Q(\overline{X}_{11} - \overline{Y}_{11})^{-1}Q^{-1}R_1^{-1}$  that  
$$L = \begin{bmatrix} 2.7857 - 25.7836 \end{bmatrix}.$$

We select the initial condition

It

$$x(0) = \begin{bmatrix} -0.3 & 0.2 & 0.02 \end{bmatrix}^{\mathrm{T}},$$

and the initial estimates are given as  $\hat{\beta}_3(0) = \hat{\beta}_4(0) = \hat{\beta}_5(0) = 0.001$ . The system state x(t) of System (3) is given in Figure 8. The sliding function s(t) is presented in Figure 9, and the control input u(t) is shown in Figure 10. The adaptive parameters are depicted in Figures 11–13.



Figure 8. State trajectories of System (1) under the adaptive SMC law.



**Figure 9.** Surface function s(t).



**Figure 10.** Adaptive SMC law u(t).



**Figure 11.** Adaptive parameter  $\hat{\beta}_3(t)$ .



**Figure 12.** Adaptive parameter  $\hat{\beta}_4(t)$ .



**Figure 13.** Adaptive parameter  $\hat{\beta}_5(t)$ .

## 5. Conclusions

In this paper, the SMC of SFOSs is studied by means of state feedback and output feedback. By designing a special sliding surface, the sliding motion of the SFOS is a normal FOS, which can be regarded as a new normalization method. A practical example and two numerical examples are utilized to prove the correctness and validity of the conclusions. The adaptive sliding mode design for Takagi-Sugeno fuzzy SFOSs by approximating the neural network is interesting and will be our future research direction.

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#### Appendix A

Figure A1 shows the part of the simulation of Example 2. The uncertainty of *A* is given in the MATLAB function module. The code of the main MATLAB function module is given as follows:

```
function dx1 = fcn(x1,x2,x3,s1,u)
A=[-1 0 -1;0 0 0;0 -1 -1];
B=[1;0;1];
U=[1;0;0];
V1=[2 4 6];
00=A+U*s1*V1;
dx1 =00(1,1)*x1+00(1,2)*x2+00(1,3)*x3+B(1,1)*(u+x1*sin(x1));
```

where OO represents the uncertainty of matrix *A*. Figure A2 shows the fractional-order model. Furthermore, Figure A2 shows the fractional module in Figure A1. In the simulation, we can adjust the initial value of the fractional order system by changing the value in the integration module.



Figure A1. Part of the simulation of Example 2.



Figure A2. Fractional-order model.

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